



# Penetration Mechanics: Analytical Modeling

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### Acknowledgements



- Virtually everything I have learned about penetration and armor mechanics occurred during my 34½ years at Southwest Research Institute
- Much of the work was funded by DARPA and/or the Army (TARDEC), although there has been funding by the Navy, Air Force, and Ballistic Missile Defense
- There also has been internal funding by SwRI



### The Charge



- Email from Clive Woodley & Ian Cullis
  - Lecture on Terminal Ballistics
    - Cover the history
    - Key developments
    - Main challenges remaining
- Decided to narrow the scope and focus on analytical modeling
  - Subject to the constraint: the model had to provide considerable insight into the mechanics of penetration
  - Major advance and not a modification of an existing model
    - Any errors of omission are solely mine (either did not think that it met the selection criteria, or ignorance on my part)
    - Any misrepresentation my responsibility

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- Rigid-body penetration Poncelet Equation
- Hydrodynamic theory
  - Assumptions
  - Results
- Modifications to hydro theory
  - Shaped-charge jets
  - Allen & Rogers
  - Chistman and Gehring





- Recht-Ipson
- Projectile deceleration Tate-Alekseevski theories
  - Target resistance
  - Cavity expansion
- Rigid-body penetration revisted
  - Tate theory
  - Forrestal, et al.'s contribution
    - Dynamic cavity expansion
    - The role of friction & temperature effects
- Transition from rigid-body to eroding penetration





- Ravid-Bodner
  - Flow fields
  - Failure mechanisms
- Walker-Anderson
  - Flow field
    - Cavity expansion
    - \* Extent of flow field
  - Effective flow stress
    - Similitude analysis





- Yarn Impact
- Fabric Modeling and Resin-Impregnated Fabrics
- Ceramics/Glasses
  - Penetration
  - Dwell
  - Dwell-penetration transition
- Summary



### Poncelet Equation



- Jean-Victor Poncelet (1788 1867)
  - French engineer and mathematician
  - Most notable work was in projective geometry
  - Commanding General of the École Polytechnique
- Poncelet equation describes rigid-body penetration

$$M\frac{d\mathbf{v}}{dt} = -F = -(A + B\mathbf{v}^2)$$

Since rigid body, non-deforming: cross-sectional area is a constant

$$\rho_p L \frac{d\mathbf{v}}{dt} = -(R_t + b\mathbf{v}^2)$$

 $R_t$  = static target resistance term

### Poncelet Equation

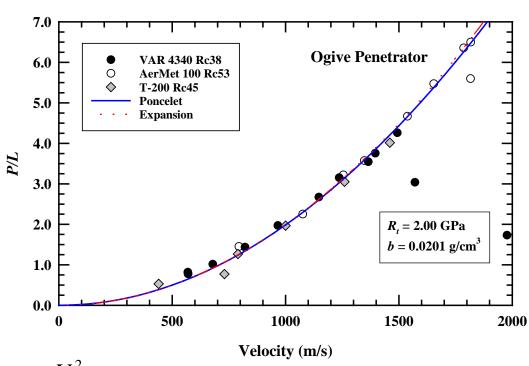


$$\frac{d\mathbf{v}}{dt} = -\frac{1}{\rho_p L} \left( R_t + b \, \mathbf{v}^2 \right)$$

$$\frac{d\mathbf{v}}{dt} = \frac{d\mathbf{v}}{dx} \frac{dx}{dt} = \mathbf{v} \frac{d\mathbf{v}}{dx}$$

$$\frac{1}{\rho_p L} \int_0^P dx = -\int_V^0 \frac{\mathbf{v} d\mathbf{v}}{R_t + b\mathbf{v}^2}$$

$$\frac{P}{L} = \frac{\rho_p}{2b} \ln \left( 1 + \frac{bV^2}{R_t} \right) \xrightarrow{small"x"} \frac{P}{L} = \frac{\rho_p V^2}{2R_t}$$



Steel into 6061 aluminum



# Tungsten Alloy into Aluminum

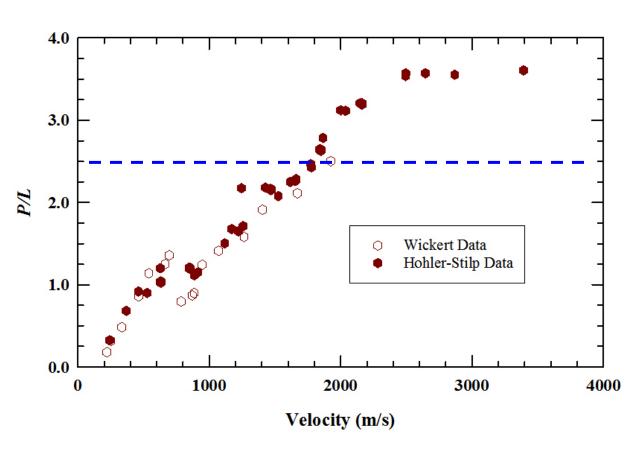


#### Flat-nosed projectiles

HS: 
$$\rho = 17.0 \text{ g/cm}^3$$
  
 $\sigma_p = 0.985 \text{ GPa}$ 

MW: 
$$\rho = 17.6 \text{ g/cm}^3$$
  
 $\sigma_p = 1.37 \text{ GPa}$ 

$$\left(\frac{P}{L}\right)_{hydro} = \left(\frac{17.3}{2.8}\right)^{1/2} = 2.48$$



A. J. Stilp and V. Hohler, "Long rod penetration mechanics," Chapter 5 in *High Velocity Impact Dynamics* (J. A. Zukas, ed.), John Wiley & Sons, NY, NY, 1990.

M. Wickert, "Penetration data for a medium caliber tungsten sinter alloy penetrator into aluminum alloy 7020 in the velocity regime from 250 m/s to 1900 m/s," *Proc.* 23<sup>rd</sup> Int. Symp. Ballistics, 2: 1437-1452, (F. Gálvez and V. Sánchez-Gálvez, Eds.), Gráficas Couche, S.L., Madrid, Spain, 2007.

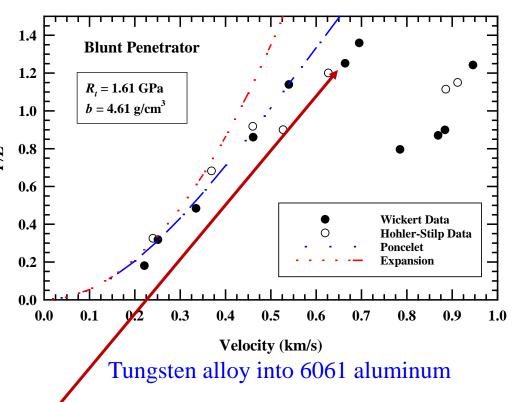
### Poncelet Equation



$$\frac{d\mathbf{v}}{dt} = -\frac{1}{\rho_p L} \left( R_t + b \, \mathbf{v}^2 \right)$$

$$\frac{P}{L} = \frac{\rho_p}{2b} \ln \left( 1 + \frac{bV^2}{R_t} \right) \xrightarrow{small"x"} \frac{P}{L} = \frac{\rho_p V^2}{2R_t}$$

Application of Poncelet solution suggests maybe not rigid-body penetration, i.e., suggestive that projectile is deforming

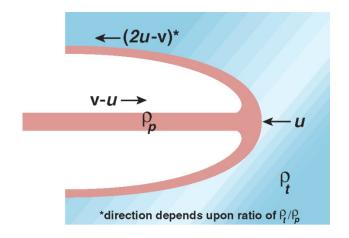


45% decrease in the fitted standard error by dropping these 3 data points





# Hydro Theory

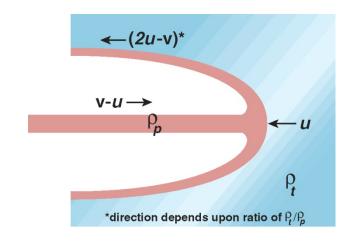


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# Consulting Hydrodynamic Theory of Penetration Swift



- First applied to shaped-charge jet penetration:
  - Developed during WWII
  - Considered a jet with constant velocity, length, and density
- Classic paper by Birkhoff, MacDougall, Pugh, and Taylor: "Explosives with lined cavities," J. Appl. Phys., 19: 563-582, 1948.
- Authors acknowledge the independent work of Hill, Mott and Pack in England that led to the same results





# Consulting Hydrodynamic Theory of Penetration Swill



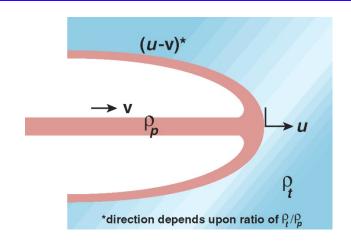
- Hydrodynamic: "To a first approximation the strengths and viscosity of target materials can be neglected and the problem can be treated by hydrodynamics."
- Incompressible jet material: "...jet with constant length  $L_o$ , velocity V, and density  $\rho_p$ "
- Incompressible target material: "penetrating a semi-infinite target of density  $\rho_t$  with a velocity u"
- Steady state: "...u has reached a constant value."

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# Inviscid Momentum Equation



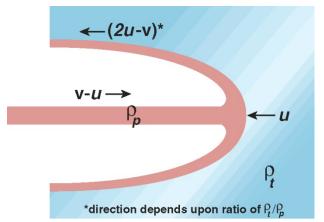


#### Hydrodynamic

$$\frac{\partial \mathbf{v}}{\partial t} + \frac{1}{2} \nabla (\mathbf{v}^2) - \mathbf{v} \times (\nabla \times \mathbf{v}) = -\frac{1}{\rho} \nabla P$$

$$\frac{1}{\rho} \nabla P = \nabla \left( \frac{P}{\rho} \right) \quad \text{Incompressible}$$

### **Change Coordination System**



$$\frac{\partial \mathbf{v}}{\partial t} + \mathbf{v} \times (\nabla \times \mathbf{v}) = \nabla \left( \frac{1}{2} \mathbf{v}^2 + \frac{P}{\rho} \right)$$

Steady state



### Momentum Equation



Take dot product of both sides with v

$$\mathbf{v} \cdot \left[ \mathbf{v} \times (\nabla \times \mathbf{v}) \right] = \mathbf{v} \cdot \nabla \left( \frac{1}{2} \mathbf{v}^2 + \frac{P}{\rho} \right)$$

 $\mathbf{v} \times (\nabla \times \mathbf{v})$  is perpendicular to  $\mathbf{v}$ 

$$\mathbf{v} \bullet \nabla \left(\frac{1}{2}\mathbf{v}^2 + \frac{P}{\rho}\right) = 0 \Rightarrow \text{ gradient of } \left(\frac{1}{2}\mathbf{v}^2 + \frac{P}{\rho}\right) \text{ is perpendicular to } \mathbf{v}$$

$$\frac{1}{2}\mathbf{v}^2 + \frac{P}{\rho} = \text{constant}$$
 hydrodynamic incompressible steady state

Very important: although has units of specific energy, this was derived from the momentum equation.



### Bernoulli Equation

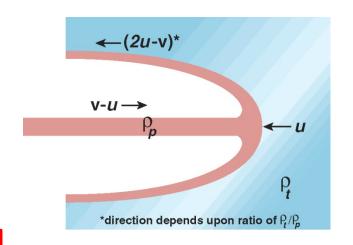


$$\frac{\partial}{\partial x} \left( \frac{1}{2} \rho \, \mathbf{v}^2 + P \right) = 0$$

#### Integrate along the centerline

$$\frac{1}{2}\rho_p(\mathbf{v}-\mathbf{u})^2 = \frac{1}{2}\rho_t \mathbf{u}^2$$

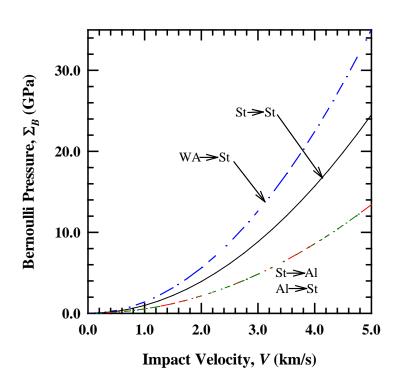
$$u = \frac{v}{1 + \sqrt{\rho_t / \rho_p}}$$

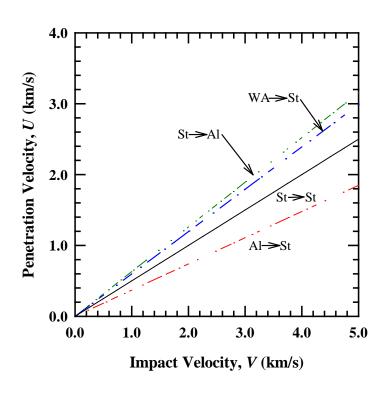




# Predictions of Theory







C. E. Anderson, Jr. and D. L. Orphal, "Re-examination of the hydrodynamic theory of penetration," *Int. J. Impact Engng.*, **35**(12): 1386-1392, 2008.



# Consulting Hydrodynamic Theory of Penetration Swill



- Additional assumptions required to derive the equation for the penetration depth:
  - Shock phase can be neglected: "...steady state is reached instantaneously"
  - No terminal phase of penetration: "...penetration stops as soon as the last particle of jet has struck the target"
- Since penetration is steady state, time of penetration:

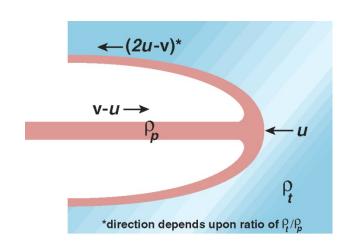
$$t = \frac{L}{v - u}$$

Penetration depth: P = ut



# Hydrodynamic Penetration Depth of Penetration





$$\frac{1}{2}\rho_p (v-u)^2 = \frac{1}{2}\rho_t u^2$$

$$u = \frac{v}{1 + \sqrt{\rho_t / \rho_p}}$$

$$P = ut t = \frac{L}{v - u}$$

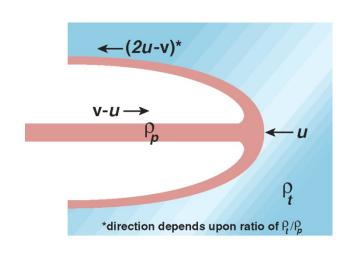
$$P = \frac{uL}{v - u}$$

$$\frac{P}{L} = \frac{u}{v - u} = \sqrt{\rho_p / \rho_t}$$



### Bernoulli Equation





$$\frac{1}{2} \rho_p (\mathbf{v} - u)^2 = \frac{1}{2} \rho_t u^2$$

$$u = \frac{\mathbf{v}}{1 + \sqrt{\rho_t / \rho_p}}$$

$$\frac{P}{L} = \sqrt{\rho_p / \rho_t}$$

Steady state Incompressible Hydrodynamic

■ 1956 Eichelberger: Need to account for target strength effects

$$\frac{1}{2} \rho_p (v-u)^2 = \frac{1}{2} \rho_t u^2 + \sigma$$

$$\sigma = \sigma_t - \sigma_p$$
  $\sigma$ : resistance to plastic deformation taken to be 1 to 3 times uniaxial yield stress



# Modified Bernoulli Model



$$\frac{1}{2} \rho_p \left( \mathbf{v} - \mathbf{u} \right)^2 = \frac{1}{2} \rho_t u^2 + \sigma$$

$$\mu = \left(\frac{\rho_t}{\rho_p}\right)^{1/2}$$

$$u = \frac{v - \mu (v^2 + 2(1 - \mu^2)\sigma/\rho_t)^{1/2}}{1 - \mu^2} \qquad \rho_p \neq \rho_t$$

$$u = \frac{\mathbf{v}}{2} - \frac{\sigma}{\rho \mathbf{v}} \qquad \rho_p = \rho_t$$



# Modified Bernoulli Equation



#### Allen & Rogers (1961)

$$\frac{1}{2} \rho_p \left( \mathbf{v} - \mathbf{u} \right)^2 = \frac{1}{2} \rho_t \mathbf{u} + \phi_t$$

- Shot 6 different projectile materials into 7075-T6 aluminum Au, Pb, Cu, Sn, Al, Mg
- Called  $\phi_t$  a dynamic yield strength of a solid target relative to a fluid jet
- $\phi_t = 3.9 Y_t$   $Y_t = 0.48 \text{ GPa}$   $\phi_t \sim 1.87 \text{ GPa}$

W. A. Allen and J. W. Rogers, "Penetration of a rod into a semi-infinite target," *J. Franklin Inst.*, **272**: 275-284, 1961.

Found the  $\phi_t$  had to be written as a function of impact velocity to reproduce final depth of penetration



# Consulting Simple Modified Bernoulli Equation Swell



$$\frac{1}{2}\rho_p \left(\mathbf{v} - \mathbf{u}\right)^2 = \frac{1}{2}\rho_t \mathbf{u}^2 + \sigma$$

- Applies to relatively weak projectiles
- Assumption: projectile is completely consumed, i.e., no projectile remains at bottom of penetration channel
- Above assumption true only for very high velocity impacts

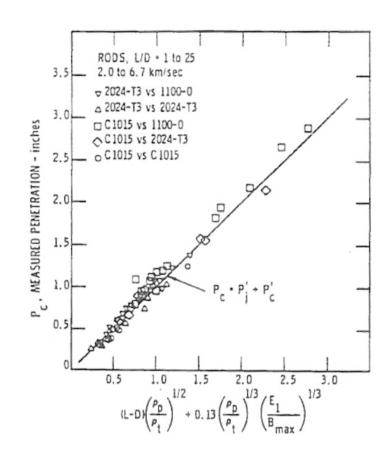


### Christman and Gehring



- Christman and Gehring described the phases of penetration for high-velocity impact
  - Separated the primary phase from the secondary (transient) phase  $P = (L-D) \left(\frac{\rho_p}{\rho_r}\right)^{1/2} + P_c$
  - $P_c$  is the crater depth obtained for a rod of L/D = 1
- Correlated  $P_c$  with experimental data  $(1 \text{ mv}^2)^{\frac{1}{3}}$

data  $P_c = 0.13 \left(\frac{\rho_p}{\rho_t}\right)^{1/3} \left(\frac{1}{2} m v^2 \frac{1}{B_{\text{max}}}\right)^{1/3}$ 



D. R. Christman and J. W. Gehring, "Analysis of high-velocity projectile penetration mechanics," *J. Appl. Phys.*, **37**(4): 1579-1587, 1966.



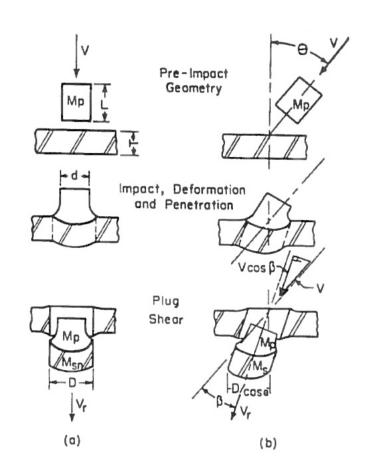
### Recht-Ipson



- Perforation of plates by blunt projectiles (fragments)
  - Normal and oblique impact
- Assumptions
  - Relatively chunky projectiles
  - Relatively thin plates

$$\left[\frac{T}{L} < \frac{1}{2}; \frac{T}{d} < \frac{1}{2}\right]$$

 Projectiles do not deform excessively (no erosion)



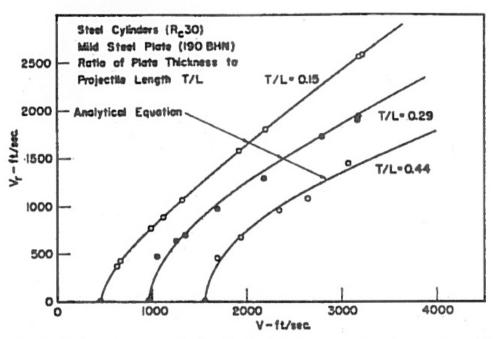
### Recht-Ipson



- Observation: a plate plug is ejected
- Conservation of momentum and energy
  - Work required to shear the plug from the target is related to V<sub>50</sub>

$$V_r = \frac{M_p}{M_p + M_{sn}} (V^2 - V_{50}^2)^{1/2}$$

$$V_{r} = \frac{1}{1 + \frac{\rho_{t}}{\rho_{p}} \left(\frac{D}{d}\right)^{2} \frac{T}{L}} \left(V^{2} - V_{50}^{2}\right)^{\frac{1}{2}}$$



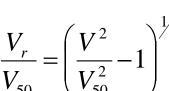
R. F. Recht and T. W. Ipson, "Ballistic perforation dynamics," *J. Appl. Mech.*, Sept.: 384-390, 1963.

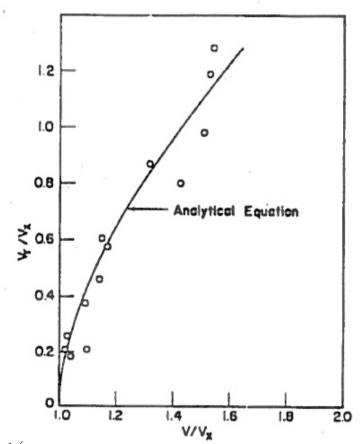


### Recht-Ipson



- Made engineering estimates for V<sub>50</sub>
- Oblique impact
  - Line-of-sight thickness
  - Angular change in fragment direction
- Thick plates perforated by cylinders
  - Problem is the plug mass; experimentally determined
- Plates perforated by AP projectiles (no plug)







# Note on Conservation of Energy



- Energy is conserved, but:
  - It is difficult to account for all the various mechanisms that dissipate energy
  - The proportion of energy dissipation (energy transfer mechanisms) changes with impact velocity
  - In particular, as the impact velocity increases, the projectile kinetic energy is transferred to the target in terms of target kinetic energy and elastic compression energy; this compression energy is dissipated by plastic work at later times\*
    - Walker demonstrated\*\* that it is the transfer of this energy at the time of penetration that defines the forces on the projectile
    - For energy rate balance to be successful, must include transfer of energy stored in the target as elastic compression
- Conservation of energy can be useful in analytical modeling, but generally over a limited velocity range and/or target-projectile configuration

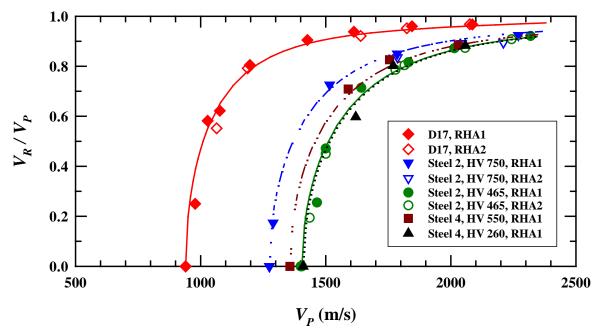
\*C. E. Anderson, Jr., D. L. Littlefield, and J. D. Walker, "Long-rod penetration, target resistance, and hypervelocity impact," *Int. J. Impact Engng.*, **14**: 1-14, 1993.

\*\*J. D. Walker, "Hypervelocity penetration modeling: momentum vs. energy and energy transfer mechanisms," *Int. J. Impact Engng.*, **26**: 809-822, 2001.

### **CEA** Consulting Residual Velocity vs. Impact Velocity Swill



- Large set of experimental data
  - Tungsten alloy and 6 different steels with different heat treats
  - 17 different projectiles
  - Two different target (armor steel) hardnesses



C. E. Anderson, Jr., V. Hohler, J. D. Walker, and A. J. Stilp, "The influence of projectile harness on ballistic performance," Int. J. Impact Engng., 22(6): 619-632, 1999.



# Scaling of Residual Velocity



$$V_{R} = \begin{cases} a(V_{p}^{m} - V_{BL}^{m})^{1/m} & V_{P} > V_{BL} \\ 0 & V_{P} \leq V_{BL} \end{cases}$$

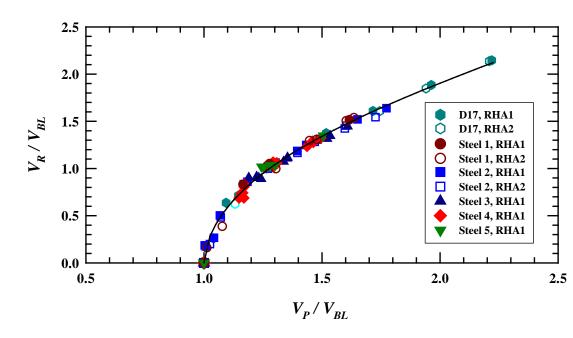
$$\frac{V_R}{V_{BL}} = a \left[ \left( \frac{V_P}{V_{BL}} \right)^m - 1 \right]^{1/m}$$



### Normalized Residual Velocity



- Similitude modeling, with correct choice of parameters, permits collapse of data
- Demonstrates that a lot of basic information about projectile and target material response is contained in V<sub>BL</sub> or V<sub>50</sub>)
- However, similitude analysis does not explicitly allow us to determine these relationships



$$\frac{V_R}{V_{BL}} = \frac{\left(0.90x^2 + 1.3x + 1.6x^{1/2}\right)}{\left(x+1\right)} \quad x = \frac{V_P}{V_{BL}} - 1$$





### Tate Model

- A. Tate, "A theory for the deceleration of long rods after impact," *J. Mech. Phys. Solids*, **15**: 387-399,1967
- A. Tate, "Further results in the theory of long rod penetration," *J. Mech. Phys. Solids*, **17**: 141-150, 1969.
- A. Tate, K. E. B. Green, P. C. Chamberlain, and R. G. Baker, "Model scale experiments on long rod penetration, *Proc. 4<sup>th</sup> Int. Symp. Ballistics*, 1978.
- A. Tate, "Long rod penetration models—Part I. A flow field model for high speed long rod penetration," *Int. J. Mech. Sci.*, **28**(8): 535-548, 1986.
- A. Tate, "Long rod penetration models—Part II. Extensions to the hydrodynamic theory of penetration," *Int. J. Mech. Sci.*, **28**(9): 599-612, 1986.
- A. Tate, "A theoretical estimate of temperature effects during rod penetration," *Proc.* 9<sup>th</sup> Int. Symp. Ballistics, **2**: 307-314, Shriverhem, UK, 1986.

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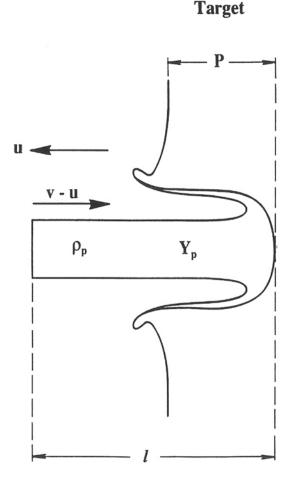
### Tate Model



- Real rods decelerate while penetrating
- Tate (1967, 1969), and Alekseevski (1966) independently postulated a different modified Bernoulli equation

$$\frac{1}{2} \rho_p (v-u)^2 + Y_p = \frac{1}{2} = \rho_t u^2 + R_t$$

- $\mathbf{Y}_{p}$ : dynamic flow stress of projectile material
- R<sub>t</sub>: target resistance



 $\rho_t$ 

R.



### Rod Deceleration



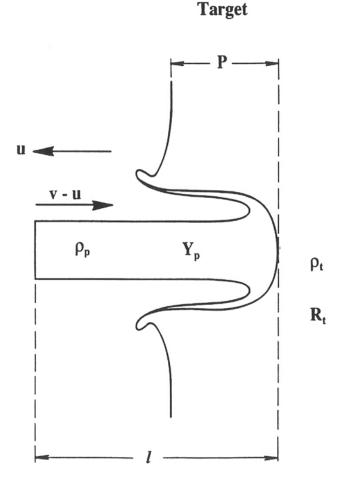
Force acting to decelerate residual rod ℓ:

$$\rho_p \ell \pi R^2 \frac{d\mathbf{v}}{dt} = -\pi R^2 Y_p$$

$$\rho_p \ell \frac{d\mathbf{v}}{dt} = -Y_p$$

Rod is getting shorter as it erodes

$$\frac{d\ell}{dt} = -\left(\mathbf{v} - \mathbf{u}\right)$$





### Tate Model

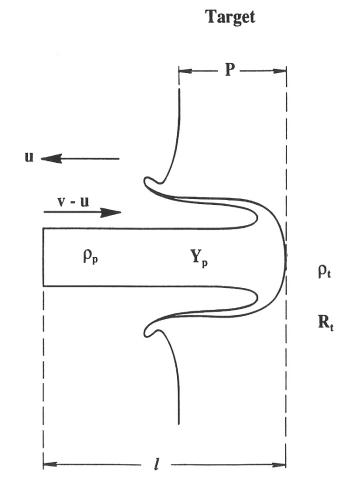


$$\frac{1}{2} \rho_p (v-u)^2 + Y_p = \frac{1}{2} \rho_t u^2 + R_t$$

$$\rho_p \ell \frac{d\mathbf{v}}{dt} = -Y_p$$

$$\frac{d\ell}{dt} = -(\mathbf{v} - \mathbf{u})$$

Simultaneous solution of three equations





## Tate Model



$$\frac{1}{2} \rho_p (v - u)^2 + Y_p = \frac{1}{2} \rho_t u^2 + R_t$$

$$u = \frac{v - \mu (v^2 + A)^{1/2}}{1 - \mu^2} \qquad \rho_p \neq \rho_t$$

$$\mu = \left(\frac{\rho_t}{\rho_p}\right)^{1/2} \quad A = \frac{2(R_t - Y_p)(1 - \mu^2)}{\rho_t}$$

$$u = \frac{\mathbf{v}}{2} - \frac{\left(R_t - Y_p\right)}{\mathbf{v} \,\rho} \qquad \rho_p = \rho_t$$

Critical Velocity (when u = 0):

$$v_c = \left[\frac{2(R_t - Y_p)}{\rho_p}\right]^{1/2}$$
 No penetration below  $v_c$ 



# How to Estimate $Y_p$ and $R_t$ ?



$$\frac{1}{2} \rho_p (v - u)^2 + Y_p = \frac{1}{2} \rho_t u^2 + R_t$$

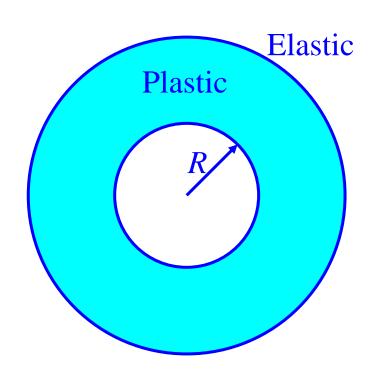
- $Y_p = (1 + \lambda)\sigma_p \qquad \sigma_p = 3.92 \bullet BHN \quad [N/mm^2]$
- $\lambda \equiv a$  factor to account for dynamic effects
- Use quasistatic cavity expansion to estimate the target resistance R<sub>t</sub>



## Cavity Expansion Theory



- Assume an incompressible or compressible plastic region and an elastic region.
- Opening up a cavity from zero radius to R quasi-statically.
- Find a similarity solution.
- Solution leads to elastic region - plastic region and allows calculation of stress at interface.



R. F. Bishop, R. Hill, and N. F. Mott, "The Theory of Indentation and Hardness," *Proc. Royal Soc.*, **57**(3): 147-159, 1945.



## Cavity Expansion Theory



Spherical Expansion (incompressible)

$$P_{sph} = \frac{2Y_t}{3} \left[ 1 + \ln \left( \frac{E_t}{(1 + v_t)Y_t} \right) \right] \rightarrow \frac{2Y_t}{3} \left[ 1 + \ln \left( \frac{2E_t}{3Y_t} \right) \right]$$

Cylindrical Expansion (incompressible)

$$P_{\text{cyl}} = \frac{Y_t}{\sqrt{3}} \left[ 1 + \ln \left( \frac{\sqrt{3} E_t}{2(1 + v_t) Y_t} \right) \right] \rightarrow \frac{Y_t}{\sqrt{3}} \left[ 1 + \ln \left( \frac{\sqrt{3} E_t}{3 Y_t} \right) \right]$$

Spherical Expansion (compressible)

$$P_{\rm sph} = \frac{2Y_t}{3} \left[ 1 + \ln \left( \frac{E_t}{3(1 - v_t)Y_t} \right) \right]$$

Cylindrical Expansion (compressible)

$$P_{\text{cyl}} = \frac{Y_t}{\sqrt{3}} \left[ 1 + \ln \left( \frac{\sqrt{3} E_t}{6(1 - v_t) Y_t} \right) \right]$$



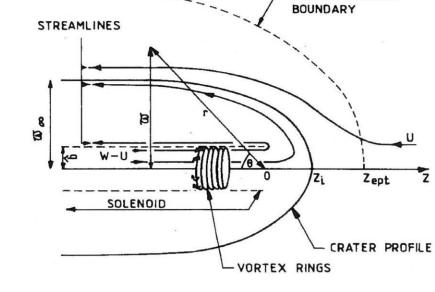
# Cavity Expansion Theory



#### Tate – Solenoidal model

- Inspired by the magnetic flow lines in a solenoid
- Material is incompressible
- The von Mises yield criterion applies
- J<sub>2</sub> flow law applies
- When yielding, the materials are perfectly plastic
- No attempt to account for rate effects, microstructural features, etc.

$$P_{\text{Tate}} = Y_t \left[ \frac{2}{3} + \ln \left( \frac{2E_t}{(4 - e^{-\lambda})Y_t} \right) \right] \qquad \frac{\lambda = 0.7}{Y_t} \left[ \frac{2}{3} + \ln \left( \frac{0.59E_t}{Y_t} \right) \right]$$



$$Y_{t} \left[ \frac{2}{3} + \ln \left( \frac{0.59 E_{t}}{Y_{t}} \right) \right]$$



# Determination of $Y_p$ and $R_t$



- $Y_p$ : dynamic yield strength of projectile  $(Y_p = 1.7\sigma_p)$
- R<sub>t</sub>: target resistance; assumed constant for a given material

$$R_{t} = \frac{2Y_{t}}{3} \left[ 1 + \ln\left(\frac{2E_{t}}{3Y_{t}}\right) \right]$$

$$R_{t} = \frac{Y_{t}}{\sqrt{3}} \left[ 1 + \ln\left(\frac{\sqrt{3}E_{t}}{3Y_{t}}\right) \right]$$

$$R_{t} = \frac{2Y_{t}}{3} \left[ 1 + \ln\left(\frac{E_{t}}{3(1-v_{t})Y_{t}}\right) \right]$$

$$R_{t} = \frac{Y_{t}}{\sqrt{3}} \left[ 1 + \ln\left(\frac{\sqrt{3}E_{t}}{6(1-v_{t})Y_{t}}\right) \right]$$

$$R_{t} = Y_{t} \left[ \frac{2}{3} + \ln \left( \frac{0.59 E_{t}}{Y_{t}} \right) \right]$$

■ *R*<sub>t</sub> is the resistance to plastic deformation



# Estimates for $R_t$



- Compute R<sub>t</sub> for an armor-like steel
  - $E_t = 200 \text{ GPa}$
  - $Y_t = 1.0 \text{ GPa}$

$$R_{t} = \frac{2Y_{t}}{3} \left[ 1 + \ln\left(\frac{2E_{t}}{3Y_{t}}\right) \right] = 3.93 \text{ GPa} \qquad R_{t} = \frac{Y_{t}}{\sqrt{3}} \left[ 1 + \ln\left(\frac{\sqrt{3} E_{t}}{3Y_{t}}\right) \right] = 3.32 \text{ GPa}$$

$$R_{t} = \frac{2Y_{t}}{3} \left[ 1 + \ln\left(\frac{E_{t}}{3(1 - v_{t})Y_{t}}\right) \right] = 3.74 \text{ GPa} \qquad R_{t} = \frac{Y_{t}}{\sqrt{3}} \left[ 1 + \ln\left(\frac{\sqrt{3} E_{t}}{6(1 - v_{t})Y_{t}}\right) \right] = 3.15 \text{ GPa}$$

$$R_{t} = Y_{t} \left[ \frac{2}{3} + \ln\left(\frac{0.59 E_{t}}{Y_{t}}\right) \right] = 5.44 \text{ GPa}$$

- R<sub>t</sub>: 3-5 times dynamic yield strength of target material
- Problem: R<sub>t</sub> is **not** a material constant; R<sub>t</sub> changes with impact velocity



## Critical Velocities



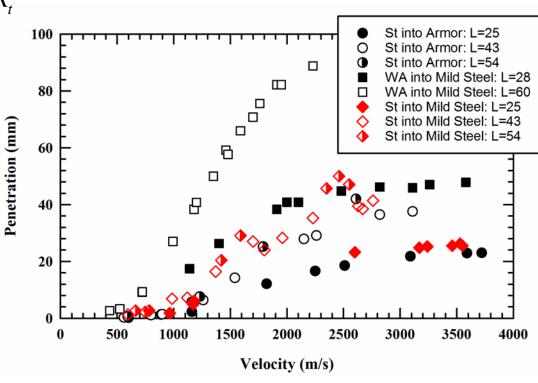
$$\frac{1}{2} \rho_p (v - u)^2 + Y_p = \frac{1}{2} \rho_t u^2 + R_t$$

- $R_t > Y_p \text{ and } u = 0$
- Critical velocity

$$\mathbf{v}_c = \left[ \frac{2(R_t - Y_p)}{\rho_p} \right]^{1/2}$$

No penetration below v<sub>c</sub>

WA into armor steel:  $v_c = 642 \text{ m/s}$ Steel into armor steel:  $v_c = 1050 \text{ m/s}$ 





## Rigid-Body Penetration



$$\frac{1}{2} \rho_p (v - u)^2 + Y_p = \frac{1}{2} \rho_t u^2 + R_t$$

- $Y_p > R_t$  and  $u \equiv v$
- Rigid-body penetration
- Crater diameter = projectile diameter





# Rigid & Eroding Penetration

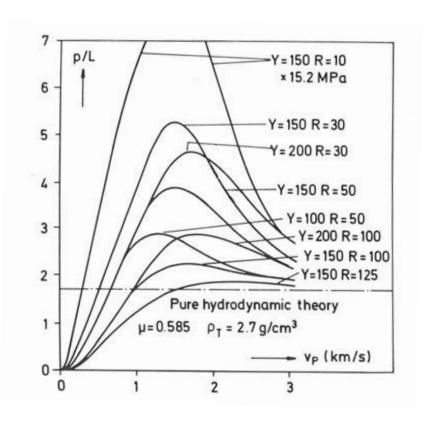


Threshold velocity

$$\mathbf{v}_{th} = \left[ \frac{2(Y_p - R_t)}{\rho_t} \right]^{1/2}$$

- $V_p < V_{th}$ , rigid
- $V_p > V_{th}$ , eroding
- Deceleration of projectile

$$\frac{d\mathbf{v}}{dt} = -\frac{\tilde{\sigma}}{\rho_p L} = -\frac{1}{\rho_p L} \left( \frac{1}{2} \rho_t \mathbf{v}^2 + R_t \right)$$



Tate model demonstrates the mechanics, but complications in trying to determine "accurate" Y<sub>p</sub> and R<sub>t</sub>



## Poncelet Equation



$$M\frac{d\mathbf{v}}{dt} = -F = -(A + B\mathbf{v}^2)$$

$$\rho_p L \frac{d\mathbf{v}}{dt} = -\left(a' + b' \,\mathbf{v}^2\right)$$

$$a' = R_t$$
  $b' = \frac{1}{2}\rho_t$  Tate

$$\frac{d\mathbf{v}}{dt} = -\frac{1}{\rho_n L} \left( R_t + \frac{1}{2} \rho_t \mathbf{v}^2 \right)$$
 "Tate" Poncelet Equation



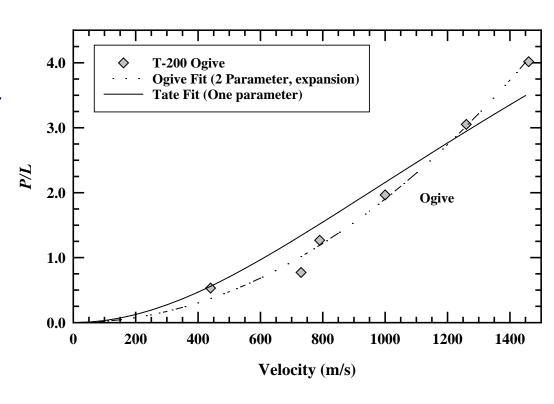
## "Tate" Poncelet



- Ogive steel penetrator into 6061-T6 aluminum
- Poncelet with 2-parameter fit; can use expansion

- $R_t = 2.10 \text{ GPa}$
- Tate: one-parameter fit

• 
$$R_t = 1.260 \text{ GPa}$$



M. J. Forrestal, J. K. Okajima, and V. K. Luk, "Penetration of 6061-T651 aluminum targets with rigid long rods," *J. Appl. Mech.* **55**: 755-760, 1988.



## Tate Model: Summary



- A one-dimensional model that predicts time history of penetration, including projectile deceleration
- Target resistance, R<sub>t</sub>, is not a constant, but depends upon material and impact velocity—although not a material property, a useful metric for comparison of different target performance
- Provides insights into dependence of penetration on material properties (e.g., density, strength)
- Predicts a critical velocity (u = 0)
- Predicts rigid-body penetration and threshold velocity for transition from rigid-body to eroding penetration



## Rigid-Body Penetration



Vascomax steel into 6061-T6 aluminum

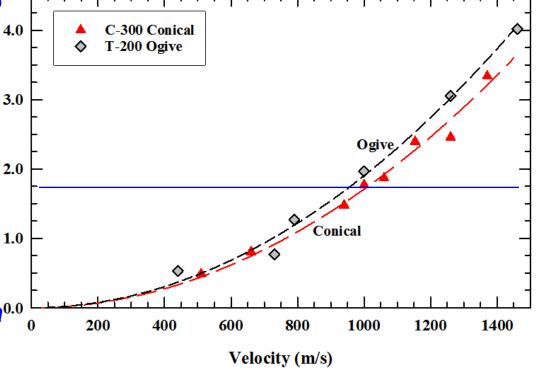
**Hydro limit** =  $\left(\frac{8.0}{2.71}\right)^{1/2}$  = 1.72

 $\frac{P}{L} = \frac{1}{2} \frac{\rho_p}{R_t} V^2$ 

 $R_t = 2.34 \text{ GPa (conical)}$ 

 $R_t = 2.10 \text{ GPa (ogive)}$ 

 $\sigma_t = 0.414 \text{ GPa}; \ \sigma_p = 1.5 \text{ GPa}^{0.0}$ 



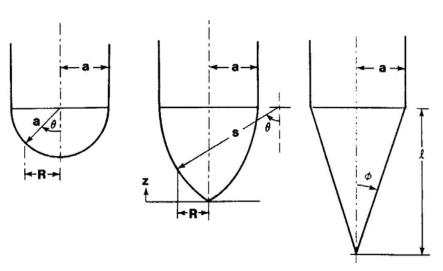
M. J. Forrestal, J. K. Okajima, and V. K. Luk, "Penetration of 6061-T651 aluminum targets with rigid long rods," *J. Appl. Mech.* **55**: 755-760, 1988.



## Forrestal & Colleagues



- Modeled rigid-body penetration
- Calculated the force on the projectile nose
  - Different nose shapes
  - Dynamic cavity expansion
  - Accurate constitutive model
    - Strain hardening
    - Rate effects



M. J. Forrestal, J. K. Okajima, and V. K. Luk, "Penetration of 6061-T651 aluminum targets with rigid long rods," *J. Appl. Mech.* **55**: 755-760, 1988.



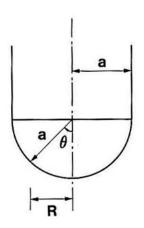
## Decelerating Force



### For the hemispherical nose:

$$dF_z = 2\pi Ra\cos\theta\sigma_n(V_z,\theta)d\theta \qquad R = a\sin\theta$$

$$F_z = \pi a^2 \int_0^{\pi/2} \sigma_n(V_z, \theta) \sin 2\theta d\theta$$



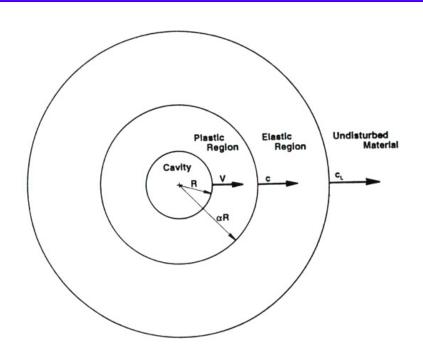
Now need to calculate  $\sigma_n(V_z, \theta)$ 



# Dynamic Cavity Expansion Theory



- Assume an incompressible or compressible plastic region and an elastic region.
- Cavity opened at constant velocity.
- Find similarity solution.
- Solution leads to elastic region - plastic region and interface velocity.
- Solution allows calculation of stress at interface.



H. G. Hopkins, "Dynamic expansion of spherical cavities in Metals," Progress in Solid Mechanics, Vol. 1 (I. Sneddon and R. Hill, Eds.), North-Holland, NY, pp. 85-164, 1960.

# Dynamic Spherical Cavity Expansion



#### Conservation of momentum

$$\frac{\partial \sigma_r}{\partial r} + \frac{2(\sigma_r - \sigma_\theta)}{r} = -\rho \left( \frac{\partial v}{\partial t} + v \frac{\partial v}{\partial r} \right)$$

#### Conservation of mass

$$\rho_o \frac{\partial}{\partial r} \left[ (r - u)^3 \right] = 3\rho r^2$$

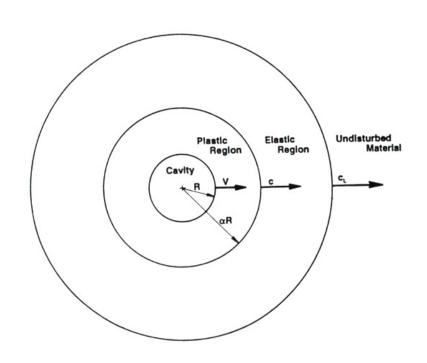
Particle displacement – particle velocity

$$\frac{\partial u}{\partial t} = v \left( 1 - \frac{\partial u}{\partial r} \right)$$

True strain rates 
$$\dot{\varepsilon}_r = -\frac{\partial v}{\partial r}$$
  $\dot{\varepsilon}_\theta = -\frac{v}{r}$ 



$$\varepsilon_r = \ln\left(1 - \frac{\partial u}{\partial r}\right)$$
  $\varepsilon_\theta = \ln\left(1 - \frac{u}{r}\right)$ 



And now a lot of math



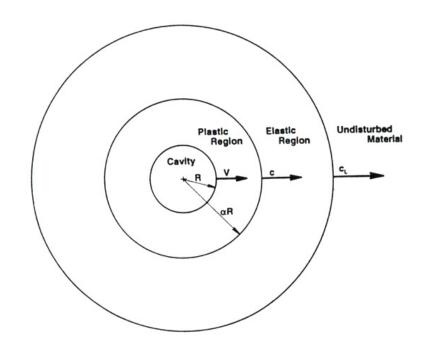
# Dynamic Spherical Cavity Expansion



$$\frac{\partial \sigma_r}{\partial r} + \frac{2(\sigma_r - \sigma_\theta)}{r} = -\rho \left( \frac{\partial v}{\partial t} + v \frac{\partial v}{\partial r} \right)$$
$$\rho_o \frac{\partial}{\partial r} \left[ (r - u)^3 \right] = 3\rho r^2$$

If incompressible:  $\rho = \rho_o$ 

$$\sigma_r = \frac{2Y_t}{3} \left[ 1 + \ln \left( \frac{2E_t}{3Y_t} \right) \right] + \frac{3}{2} \rho_t V^2 = A + BV^2$$



Now have a resistance term that includes strength and inertial effects

Also note that the "resistance" is proportional to the square of the cavity velocity

Note: in virtually all the work of Forrestal and colleagues, they used spherical cavity expansion



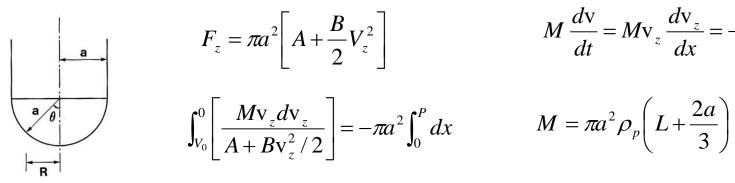
## Decelerating Force



#### For the hemispherical nose:

$$F_z = \pi a^2 \int_0^{\pi/2} \sigma_n(V_z, \theta) \sin 2\theta d\theta$$

$$\sigma_n(V_z, \theta) = A + BV^2 = A + BV_z^2 \cos^2 \theta$$



$$F_z = \pi a^2 \left[ A + \frac{B}{2} V_z^2 \right]$$

$$\int_{V_0}^0 \left[ \frac{M \mathbf{v}_z d \mathbf{v}_z}{\Delta + R \mathbf{v}^2 / 2} \right] = -\pi a^2 \int_0^P dx$$

$$M\frac{d\mathbf{v}}{dt} = M\mathbf{v}_z \frac{d\mathbf{v}_z}{dx} = -F_z$$

$$M = \pi a^2 \rho_p \left( L + \frac{2a}{3} \right)$$

$$\frac{P}{L+2a/3} = \frac{\rho_p}{B} \ln \left(1 + \frac{BV_0^2}{2A}\right) = \frac{2\rho_p}{3\rho_t} \ln \left(1 + \frac{\frac{3}{2}\rho_t V_0^2}{\frac{4}{3}Y_t \left[1 + \ln \left(\frac{2E}{3Y_t}\right)\right]}\right)$$



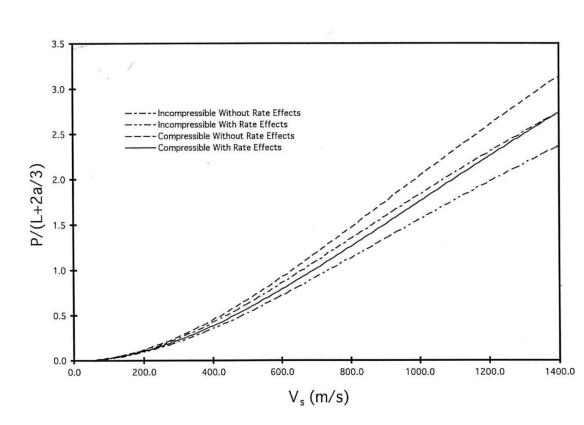
## Accurate Constitutive Modeling



$$\frac{\partial \sigma_r}{\partial r} + \frac{2(\sigma_r - \sigma_\theta)}{r} = -\rho \left( \frac{\partial v}{\partial t} + v \frac{\partial v}{\partial r} \right)$$

$$\sigma = \begin{cases} E\varepsilon & \sigma \leq Y_d & \widehat{\varphi}_{2.0} \\ Y\left(\frac{E\varepsilon}{Y_d}\right)^n + \alpha \left(\frac{\dot{\varepsilon}}{\dot{\varepsilon}_o}\right)^m & \sigma > Y_d & \stackrel{+}{\searrow} \\ 1.5 \end{cases}$$

$$Y_{d} = Y + \alpha \left(\frac{\dot{\varepsilon}}{\dot{\varepsilon}_{o}}\right)^{m}$$



#### And now a whole lot of math

T. L. Warren and M. J. Forrestal, "Effects of strain hardening and strain-rate sensitivity on the penetration of aluminum targets with spherical-nosed rods," *Int. J. Solids Structures* **35**(28-29): 3737-3753, 1998.

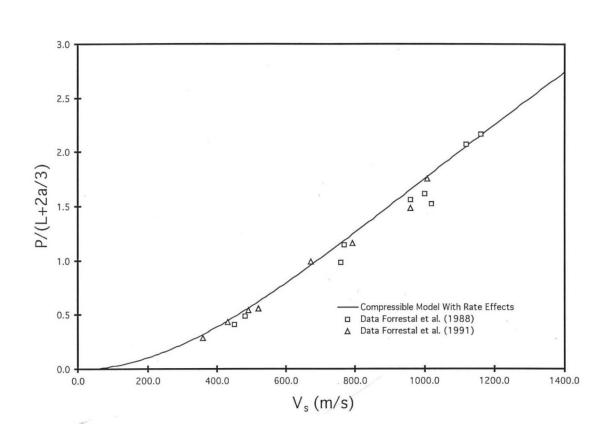


## Accurate Constitutive Modeling



$$\sigma = \begin{cases} E\varepsilon & \sigma \leq Y_d \\ Y \left(\frac{E\varepsilon}{Y_d}\right)^n + \alpha \left(\frac{\dot{\varepsilon}}{\dot{\varepsilon}_o}\right)^m & \sigma > Y_d \end{cases}$$

$$Y_d = Y + \alpha \left(\frac{\dot{\varepsilon}}{\dot{\varepsilon}_o}\right)^m$$



T. L. Warren and M. J. Forrestal, "Effects of strain hardening and strain-rate sensitivity on the penetration of aluminum targets with spherical-nosed rods," *Int. J. Solids Structures* **35**(28-29): 3737-3753, 1998.



## Friction



- Initially, Forrestal and colleagues were concerned about friction effects between the projectile and penetration cavity wall included frictional forces in their model
- Later, determined that when used accurate constitutive model, there was no need to include friction



## Camacho and Ortiz



- Performed detailed finite element simulations of some of the experiments, using a new adaptive meshing technique and a constitutive material law
  - strain hardening
  - rate-dependent plasticity
  - heat conduction
  - thermal-mechanical coupling
- Simulations showed a very thin melted layer in the target next to the projectile that resulted in a nearly frictionless interface

G.T. Camacho and M. Ortiz, "Adaptive Lagrangian modeling of ballistic penetration of metallic targets," *Comput. Methods Appl. Mech. Eng.*, 142:269-301, 1997.



## Temperature Effects



- Tate made an estimate of temperature effects during rod penetration
  - Thermal conduction is significant only very close to the interface
  - When distances are scaled relative to the crater diameter:
    - Temperature distribution is independent of the impact velocity
    - Temperature approaches the melting temperature in a small region which is of the same order of size as the conduction dominated zone

A. Tate, "A theoretical estimate of temperature effects during rod penetration," *Proc.* 9<sup>th</sup> Int. Symp. Ballistics, 2-307-314, Shriverham, UK, 1996.



## Friction Effects



- Mark Wilkins initially did not have friction forces in Lagrangian hydrocode HEMP, but had good agreement with experiments
- Added friction, and calculated results got worse
- Friction in penetration mechanics: that quantity added to an analytic model to improve agreement with experiment, whose sole justification is that the friction coefficient is on the order of 0.1 0.2 and thus appears reasonable





# Rigid-Body to Eroding Penetration

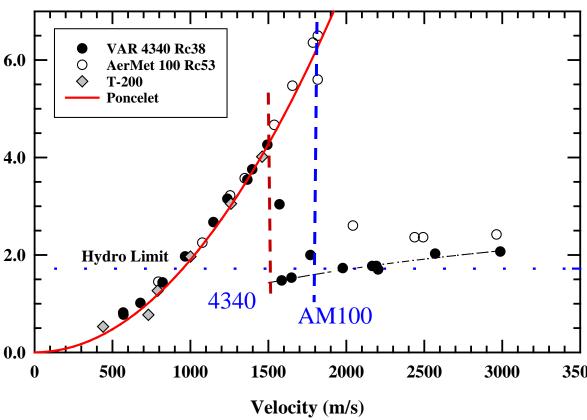


## High-Velocity Ogive Data



#### Projectile strength

- T-200:  $\sigma_p = 1.38$  GPa
- 4340 VAR:  $\sigma_p = 1.50$  GPa
- AerMet100:  $\sigma_p = 1.93$  GPa
- Transition occurs over very narrow velocity interval



A. J. Piekutowski, M. J. Forrestal, K. L. Poormon, and T. L. Warren, "Penetration of 6061-T6511 aluminum targets by ogive-nose steel projectiles with striking velocities between 0.5 and 3.0 km/s," *Int. J. Impact Engng.*, **23**: 723-734, 1999.



## Tungsten Alloy into Aluminum

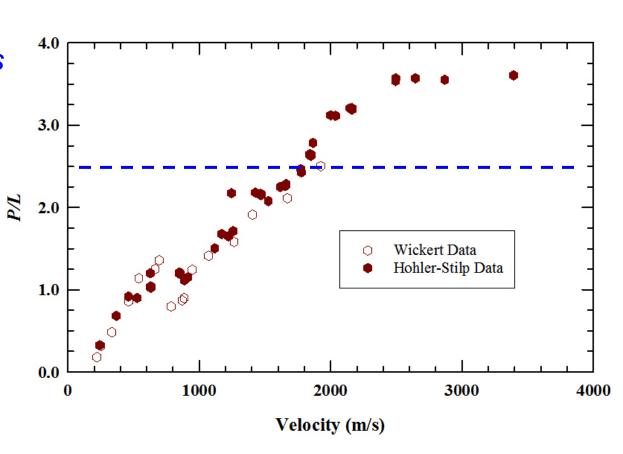


### Flat-nosed projectiles

HS: 
$$\rho = 17.0 \text{ g/cm}^3$$
  
 $\sigma_p = 0.985 \text{ GPa}$ 

MW: 
$$\rho = 17.6 \text{ g/cm}^3$$
  
 $\sigma_p = 1.37 \text{ GPa}$ 

$$\left(\frac{P}{L}\right)_{hydro} = \left(\frac{17.3}{2.8}\right)^{1/2} = 2.48$$



A. J. Stilp and V. Hohler, "Long rod penetration mechanics," Chapter 5 in *High Velocity Impact Dynamics* (J. A. Zukas, ed.), John Wiley & Sons, NY, NY, 1990.

M. Wickert, "Penetration data for a medium caliber tungsten sinter alloy penetrator into aluminum alloy 7020 in the velocity regime from 250 m/s to 1900 m/s," *Proc.* 23<sup>rd</sup> Int. Symp. Ballistics, 2: 1437-1452, (F. Gálvez and V. Sánchez-Gálvez, Eds.), Gráficas Couche, S.L., Madrid, Spain, 2007.

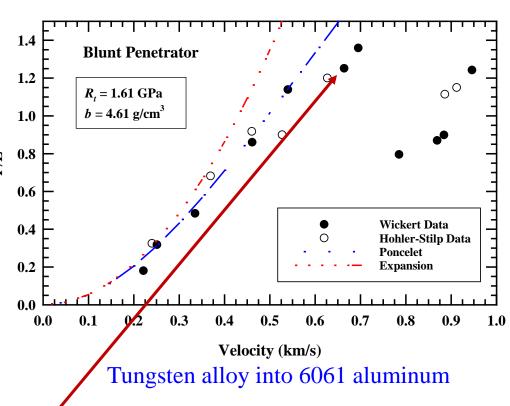
## Poncelet Equation



$$\frac{d\mathbf{v}}{dt} = -\frac{1}{\rho_p L} \left( R_t + b \, \mathbf{v}^2 \right)$$

$$\frac{P}{L} = \frac{\rho_p}{2b} \ln \left( 1 + \frac{bV^2}{R_t} \right) \xrightarrow{small"x"} \frac{P}{L} = \frac{\rho_p V^2}{2R_t}$$

Tungsten alloy more ductile than the very hard steels, allowing mushrooming before eroding

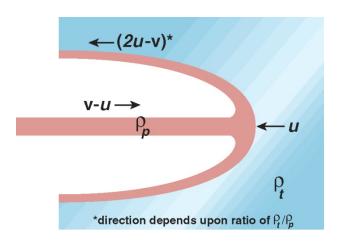


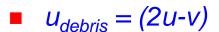
45% decrease in the fitted standard error by dropping these 3 data points

CEA
Consulting

# Tungsten Alloy into Aluminum Secondary Penetration



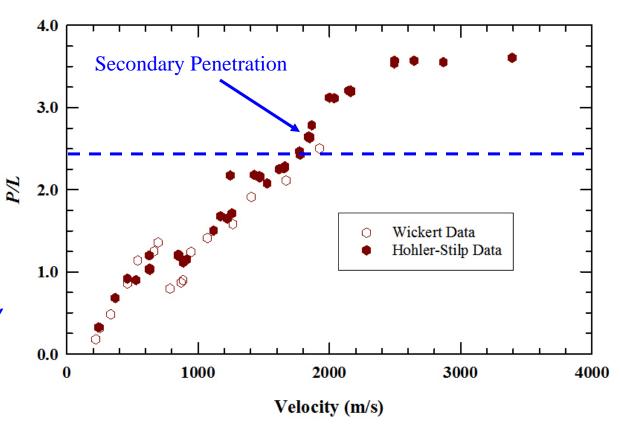




$$\rho_p << \rho_t$$
,  $u \approx 0$ ,  $u_{debris} \sim -v$ 

$$\rho_p = \rho_t, \ U_{debris} = 0$$

•  $\rho_p > \rho_t$ ,  $u_{debris}$  into target





## Rigid-Body Penetration



- Can obtain deep penetration when projectile remains rigid, particularly compared to eroding penetration
- Diameter of penetration channel is the diameter of the projectile
- At sufficiently high velocities, projectile begins to deform and then begins to erode
  - Deformation occurs over a relatively small range of impact velocities
  - Projectile material strength and ductility important near/at transition velocity
  - Nose shape is important
- Projectile typically does not want to penetrate straight





# Dynamic Plasticity

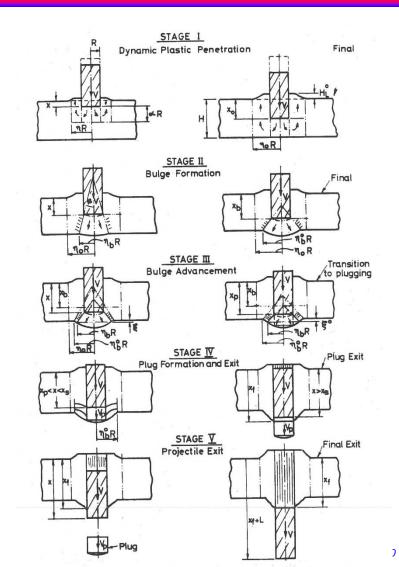
Plastic Deformation Flow Fields



## Ravid-Bodner Model



- Multi-stage penetration/perforation model by a rigid projectile
  - Penetration
  - Bulge formation
  - Bulge advancement
  - Plug formation and exit
  - Projectile exit



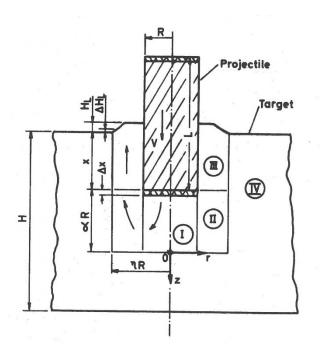
M. Ravid and S. R. Bodner, Dynamic perforation of viscoplastic plates by rigid projectiles," *Int. J. Engng. Sci.*, **21**(6): 577-591, 1983.



## Ravid-Bodner Model



- Established plastic flow fields within the target
  - Specified velocity distributions subject to compatibility and continuity conditions
  - Computed plastic work rates (including strain-rate effects)
- Solved for the radial and axial extent of the plastic zone fields (ηR and αR, where R is the projectile radius)
- Deceleration of projectile computed from an energy rate balance



M. Ravid and S. R. Bodner, Dynamic perforation of viscoplastic plates by rigid projectiles," *Int. J. Engng. Sci.*, **21**(6): 577-591, 1983.

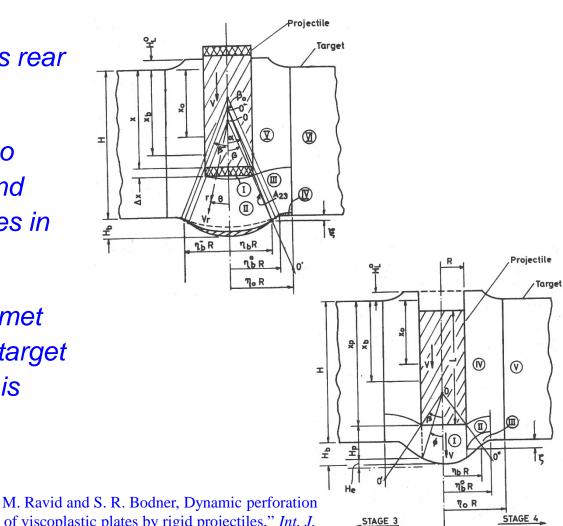


## Ravid-Bodner Model



72

- When plastic zone reaches rear of target, bulging begins
- At some point, the bulge no longer expands radially, and instead, the bulge advances in direction of penetration
- When a failure criterion is met (variety of failure modes), target material fails and the plug is ejected
- Projectile exits the target

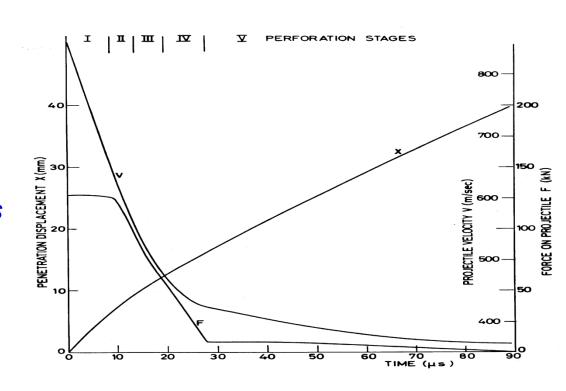




### Ravid-Bodner Model



- Steel plate: 12-mm thick
- 7.62-mm AP bullet
- Impact velocity: 855 m/s
- Exit velocity: 300-390 m/s
- Model
  - Plug velocity: 424 m/s
  - Bullet velocity: 364 m/s



M. Ravid and S. R. Bodner, Dynamic perforation of viscoplastic plates by rigid projectiles," *Int. J. Engng. Sci.*, **21**(6): 577-591, 1983.



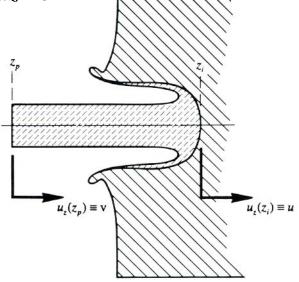
### Walker-Anderson Model



$$\rho_{p} \int_{z_{p}}^{z_{1}} \frac{\partial u_{z}}{\partial t} dz + \rho_{t} \int_{z_{i}}^{+\infty} \frac{\partial u_{z}}{\partial t} dz + \frac{1}{2} \rho_{p} u_{z}^{2} \bigg|_{z_{p}}^{z_{i}} + \frac{1}{2} \rho_{t} u_{z}^{2} \bigg|_{z_{i}}^{+\infty} - \sigma_{zz} \bigg|_{z_{p}}^{+\infty} - 2 \int_{z_{p}}^{+\infty} \frac{\partial \sigma_{xz}}{\partial x} dz = 0$$

$$\rho_{p} \int_{z_{p}}^{z_{i}} \frac{\partial u_{z}}{\partial t} dz + \rho_{t} \int_{z_{i}}^{+\infty} \frac{\partial u_{z}}{\partial t} dz + \frac{1}{2} \rho_{p} \left(u^{2} - v^{2}\right) - \frac{1}{2} \rho_{t} u^{2} - \int_{z_{p}}^{+\infty} \frac{\partial \sigma_{xz}}{\partial x} dz = 0$$

- Integrate momentum equation along the centerline over the projectile and target
- Three assumptions
  - Velocity profile along the centerline in projectile and target specified
  - The rear of the projectile is decelerated by elastic waves
  - Shear behavior of the target material is specified



J. D. Walker and C. E. Anderson, Jr., "A time-dependent model for long-rod penetration," *Int. J. Impact Engng.*, **16**(1): 19-48, 1995.

#### **CEA**

# Consulting Velocity Profile Along the Centerline Swift

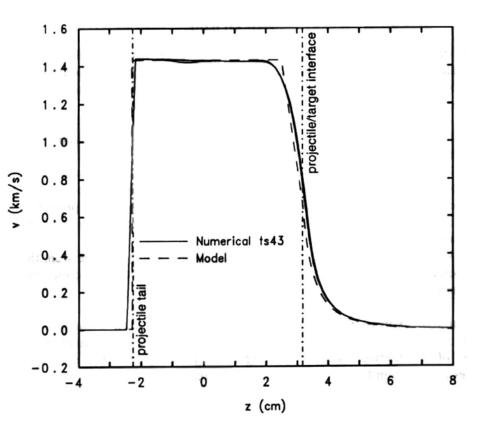


#### Projectile

$$u_{z}(z) = \begin{cases} u - \frac{v - u}{s} (z - z_{i}) & (z_{i} - s) \leq z < z_{i} \\ v & z_{p} \leq z < (z_{i} - s) \end{cases}$$

#### **Target**

$$u_{z}(z) = \begin{cases} \frac{u}{\alpha^{2} - 1} \left[ \left( \frac{\alpha R}{r(z)} \right)^{2} - 1 \right] & R \le r(z) < \alpha R \\ 0 & r(z) \ge \alpha R \end{cases}$$



Velocity field derived from a vector potential

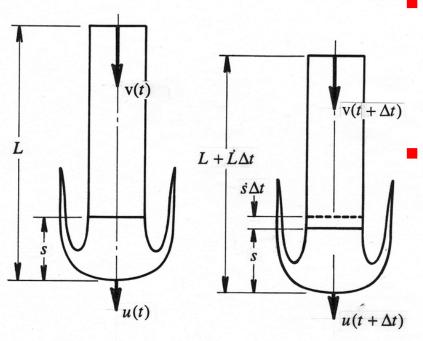
Extent of flow field in target =  $\alpha R$ R = crater radius

J. D. Walker and C. E. Anderson, Jr., "A timedependent model for long-rod penetration," Int. J. Impact Engng., 16(1): 19-48, 1995.



### Projectile Deceleration





Time:  $t + \Delta t$ 

J. D. Walker and C. E. Anderson, Jr., "A time-dependent model for long-rod penetration," *Int. J. Impact Engng.*, **16**(1): 19-48, 1995.

Time: t

Front end and back end of projectile have different velocities - thus projectile erodes during penetration event.

$$\dot{L} = -(v - u)$$

Back of projectile is decelerated by elastic waves that travel up and down the length of the projectile, reflecting off the free surface at the back and at the elastic-plastic interface at the front.

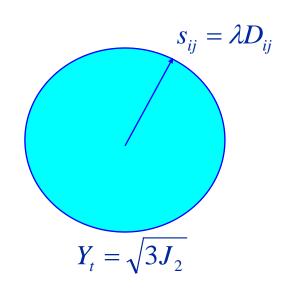
$$\Delta \mathbf{v} = -2c \frac{\sigma_p}{E_p}$$

$$\dot{\mathbf{v}} = -\frac{\sigma_p}{\rho_p(L-s)} \left\{ 1 + \frac{\mathbf{v} - u}{c} + \frac{\dot{s}}{c} \right\}$$



### Shear Behavior of Target





- Shear stress is directly proportional to rate of deformation (rigid plasticity)  $s_{ii} = \lambda D_{ii}$
- A von Mises yield surface is assumed

$$s_{ij}s_{ij} = 2J_2 = \frac{2}{3}Y_t^2$$
 or  $Y_t = \sqrt{3J_2}$   
 $s_{ij} = \lambda D_{ij} = \frac{\sqrt{\frac{2}{3}}Y_t}{\sqrt{D_{ij}D_{ij}}}D_{ij}$ 

Assumed target flow field gives

$$\alpha = \frac{Extent of \ Plastic \ Zone}{Crater \ Radius}$$

$$\left. \int_{R}^{\alpha R} \frac{\partial s_{xz}}{\partial x} \right|_{x=0} dz = -\frac{7Y_{t}}{6} \ln(\alpha)$$



### Walker-Anderson Model



$$\rho_{p}\dot{\mathbf{v}}(L-s) + \dot{\mathbf{u}}\left\{\rho_{p}s + \rho_{t}R\frac{\alpha - 1}{\alpha + 1}\right\} + \rho_{p}\frac{d}{dt}\left(\frac{\mathbf{v} - \mathbf{u}}{s}\right)\frac{s^{2}}{2} + \rho_{t}\dot{\alpha}\frac{2R\mathbf{u}}{(\alpha + 1)^{2}}$$

$$= \frac{1}{2}\rho_{p}(\mathbf{v} - \mathbf{u})^{2} - \left\{\frac{1}{2}\rho_{t}\mathbf{u}^{2} + \frac{7}{3}Y_{t}\ln\alpha\right\} \quad \text{(Centerline momentum balance)}$$

$$\dot{\mathbf{v}} = -\frac{\sigma_p}{\rho_p(L-s)} \left\{ 1 + \frac{\mathbf{v} - u}{c} + \frac{\dot{s}}{c} \right\}$$

(Projectile rear deceleration)

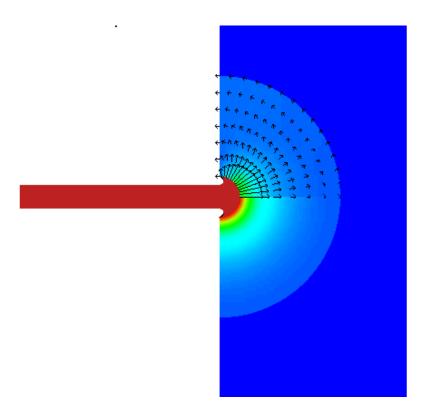
$$\dot{L} = -(\mathbf{v} - u)$$
 (Erosion)

J. D. Walker and C. E. Anderson, Jr., "A time-dependent model for long-rod penetration," *Int. J. Impact Engng.*, **16**(1): 19-48, 1995.



### Target Resistance





Ignoring transient terms, the target resistance is

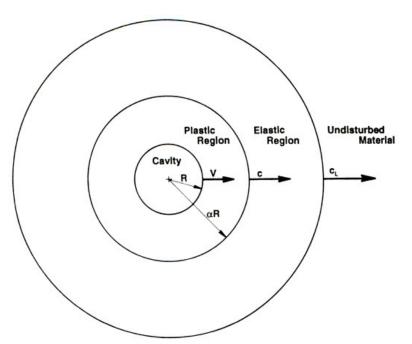
target resistance ~ 
$$-\left\{\frac{1}{2}\rho_{t}u^{2} + \frac{7Y_{t}}{3}\ln(\alpha)\right\}$$

the first term is due to inertial terms, moving target material out of the way, and the second term is due to plastic flow in the target.

## **CEA**

# Consulting Estimating the Extent of Plastic Flow Swell





2-D Cylindrical Cavity Expansion

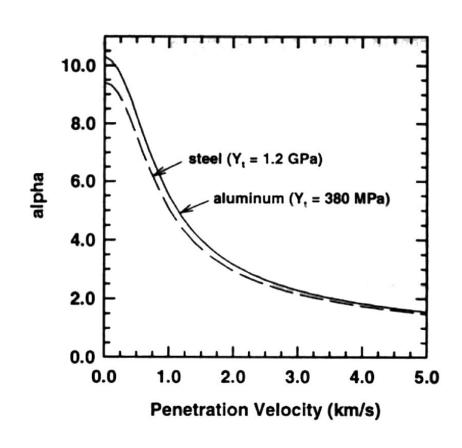
S. Chocron, C. E. Anderson, Jr., and J. D. Walker, "Long-rod penetration: Cylindrical vs. spherical cavity expansion for the extent of plastic flow," *Proc 17th Symp. Ballistics*, 1998.

- Cylindrical compressible cavity expansion
- Two regions identified, separated by three boundaries with appropriate B.C.:
  - elastic
  - elastic-plastic (compressible)
- The inner cavity is driven at a constant velocity V
- Important approximate assumption: the velocity is assumed continuous between the elastic and elastic-plastic region
  - Cylindrical cavity expansion considerably more accurate than spherical cavity expansion



# 





 $\alpha$  is a function of impact velocity

- Interface between elastic-plastic and elastic region moves at a constant velocity c.
- The extent of the plastic flow field within the target is defined as

$$\alpha(u) = c(u)/u$$

The cavity expansion provides an expression for this extent:

$$\left(1 + \frac{\rho_t u^2}{Y_t}\right) \sqrt{K_t - \rho_t u^2 \alpha} = \left(1 + \frac{\rho_t u^2 \alpha^2}{2G_t}\right) \sqrt{K_t - \rho_t u^2}$$

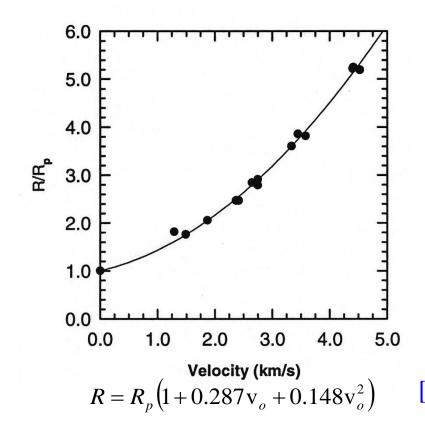
- Compressibility of the target is reflected in  $\alpha$ .
- High pressure stiffness adjustment

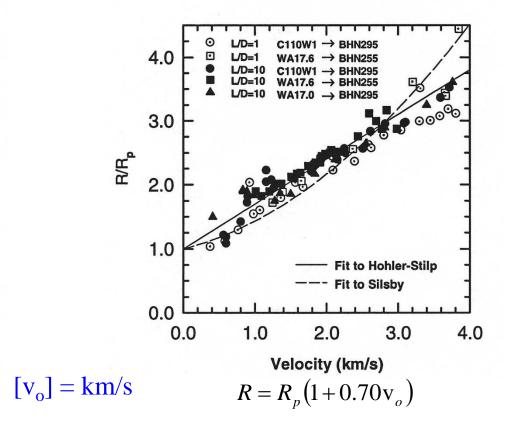
$$K_t \sim K_0 \left( 1 + k \frac{u_p}{c_0} \right)$$

#### Crater Radius



 Curve fit based on experimental data: depends on projectile and target material properties and penetration velocity







### Walker-Anderson Model Results

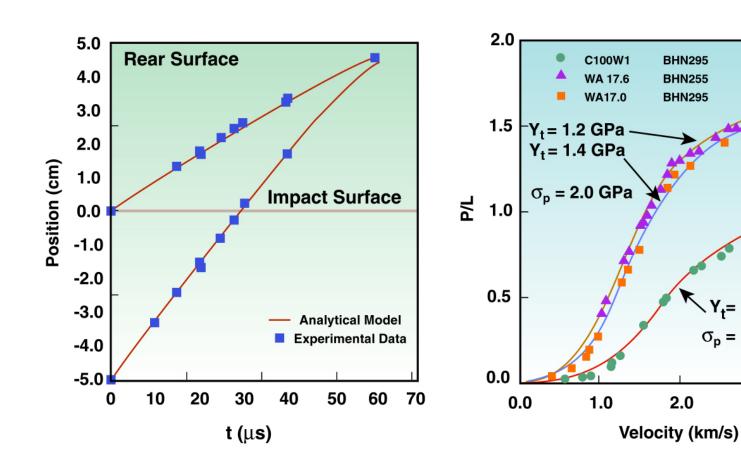


Y<sub>t</sub>= 1.4 GPa

 $\sigma_p = 1.2 \text{ GPa}$ 

3.0

4.0



J. D. Walker and C. E. Anderson, Jr., "A time-dependent model for long-rod penetration," *Int. J. Impact Engng.*, **16**(1): 19-48, 1995.



# Walker-Anderson Model ⇒ Tate Model



$$\rho_{p}\dot{\mathbf{v}}(L-s) + \dot{u}\left\{\rho_{p}s + \rho_{t}R\frac{\alpha - 1}{\alpha + 1}\right\} + \rho_{p}\frac{d}{dt}\left(\frac{\mathbf{v} - \mathbf{u}}{s}\right)\frac{s^{2}}{2} + \rho_{t}\dot{\alpha}\frac{2R\mathbf{u}}{(\alpha + 1)^{2}}$$

$$= \frac{1}{2}\rho_{p}(\mathbf{v} - \mathbf{u})^{2} - \left\{\frac{1}{2}\rho_{t}\mathbf{u}^{2} + \frac{7}{3}Y_{t}\ln\alpha\right\}$$

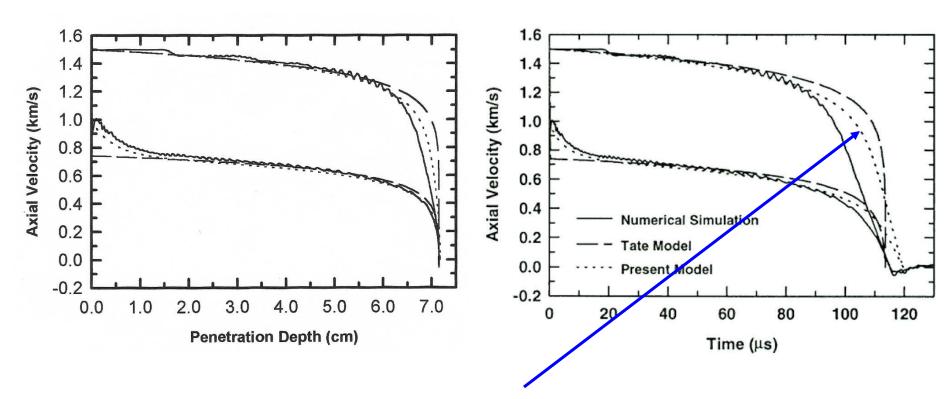
■ In the limit where the three dimensional terms are removed, the Walker-Anderson penetration model reduces to the Tate model:  $R \rightarrow 0$  and  $s \rightarrow 0$ 

Also let the Young's modulus in the projectile become large:  $c \rightarrow \infty$ 

Then 
$$-\rho_p \dot{\mathbf{v}} L + \frac{1}{2} \rho_p (\mathbf{v} - \mathbf{u})^2 = \frac{1}{2} \rho_t \mathbf{u}^2 + \frac{7Y_t}{3} \ln(\alpha)$$
$$\dot{\mathbf{v}} = -\frac{\sigma_p}{\rho_p L} \qquad \dot{L} = -(\mathbf{v} - \mathbf{u}) \qquad \text{Tate:} \qquad Y_p = 1.7\sigma_p$$
$$R_t = \frac{7Y_t}{3} \ln(\alpha)$$

### Walker-Anderson Model Results



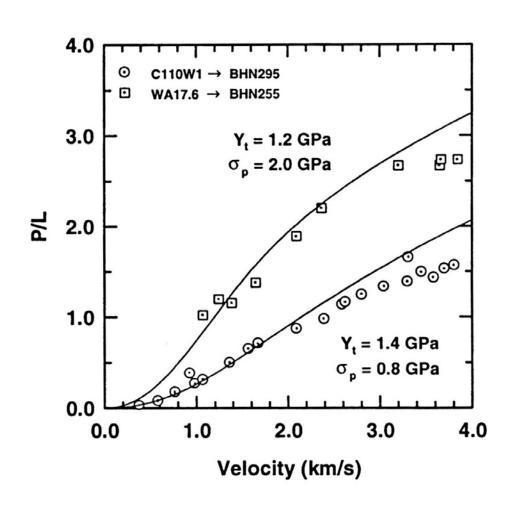


When rod is short ( $L/D \sim 1$ ), tail of rod does not decelerate quite as rapidly as should; tail "sees" the high-pressures at projectile-target interface, not just elastic deceleration waves



### Walker-Anderson Model Results

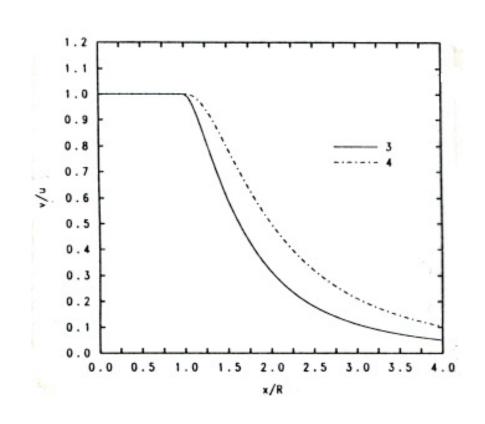




L/D = 1

### Bulging: Shear Flow Field





J. D. Walker, "An analytic velocity field for back surface bulging," *Proc. Int. Symp. Ballistics*, **2**: 1239-1246,

Technomic Publishing Company, Lancaster, PA, 1999.

For back surface bulge, a flow field is achieved through a multiplicative blending of the potentials for hemispherical flow and a shear flow

$$\vec{A}(r,\theta) = A_{deep}^{\lambda} A_{shear}^{1-\lambda} \hat{e}_{\phi}$$

Velocity is obtained from curl

$$\vec{\mathbf{v}} = curl(\mathbf{A}(r,\theta)\hat{e}_{\phi})$$

Two shear flows are

$$(\mathbf{v}_z)_3 = u \left(\frac{R}{x}\right)^3 \left\{4 - 3\frac{R}{x}\right\}$$

$$(\mathbf{v}_z)_4 = u \left(\frac{R}{x}\right)^3 \left\{ 10 - 15\frac{R}{x} + 6\left(\frac{R}{x}\right)^2 \right\}$$

### **CEA** Consulting Modifications for Back Surface Bulge Swill



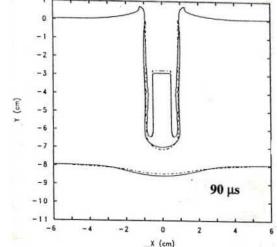
Amount of material with shear vs. deep penetration potential is based on the hemisphere volume (radius R) overlapping the remaining thickness cylinder volume (thickness T)

$$\lambda = \frac{3}{2} \frac{T}{\widetilde{R}} - \frac{1}{2} \left( \frac{T}{\widetilde{R}} \right)^3$$

Target resistance term is replaced by

$$-\left\{\frac{1}{2}\rho_t u^2 + \frac{7}{3}Y_t \ln(\alpha)\right\} \quad \text{by} \quad -\frac{1}{2}\rho_t \left(u - u_{back}\right)^2 + \frac{7}{3}Y_t \ln(\widetilde{\alpha})$$

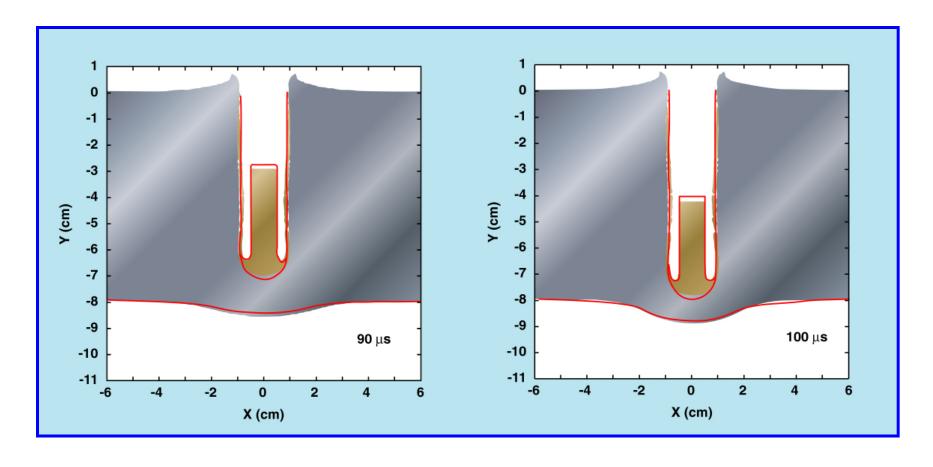
Back surface location can be computed since the velocity is known.





# Analytical Model Compared to Numerical Simulations



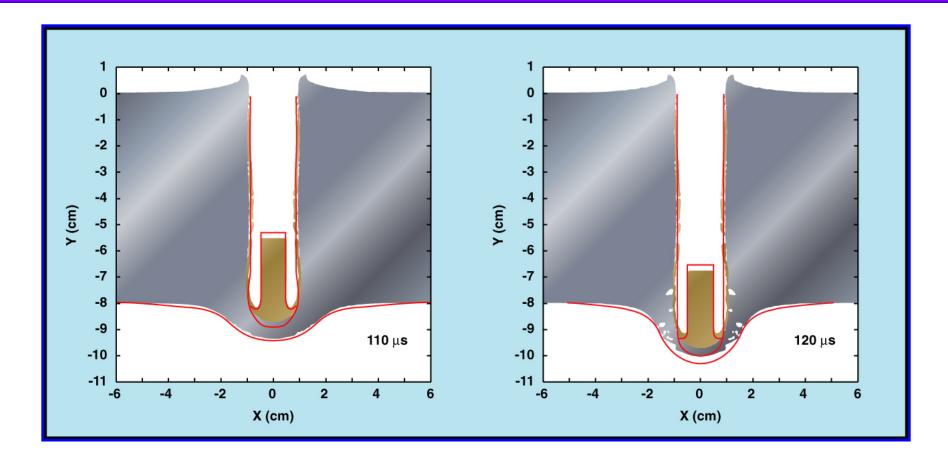


Tungsten into Steel, L/D=10, 1.5 km/s



# Analytical Model Compared to Numerical Simulations



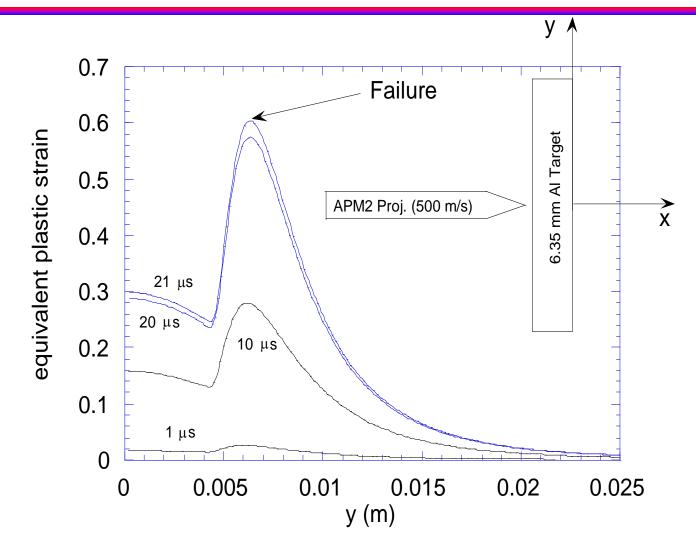


Tungsten into Steel, L/D=10, 1.5 km/s



### Back Surface Strains

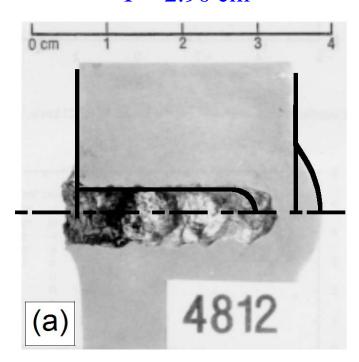




# Analytical Model Compared to Experiment

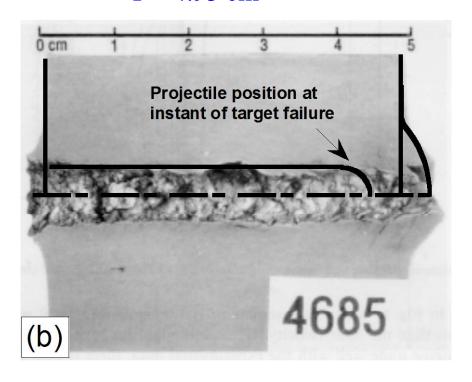


T = 2.90 cm



$$V_0 = 1241 \text{ m/s}$$

T = 4.95 cm



$$V_0 = 1700 \text{ m/s}$$



### Model Results







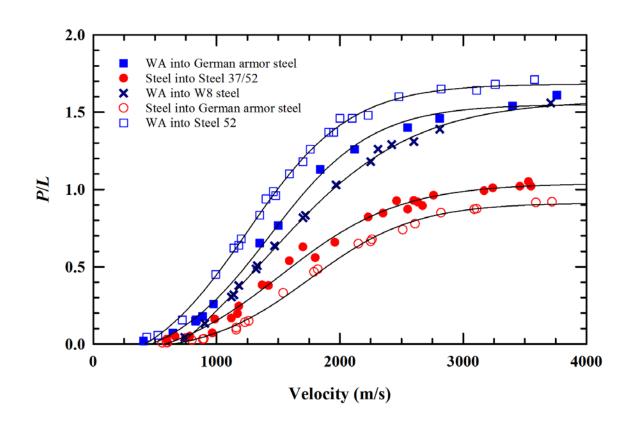


## Similitude Analysis



### P/L vs. Impact Velocity





Data from Hohler and Stilp, compiled in: C. E. Anderson, Jr., B. L. Morris, and D. L. Littlefield, "A penetration mechanics database," SwRI Report 3593/001, San Antonio TX, 1992, prepared for DARPA.



### Effective Flow Stress

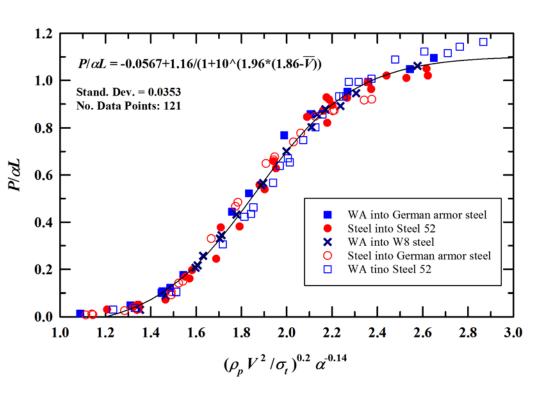


	$\rho$ (g/cm <sup>3</sup> )	$\sigma_t$ (GPa-)
tungsten alloy	17.0	-
W8 steel	7.85	1.57
St 52 steel	7.85	0.858
St37/52 steel	7.85	1.09
German armor steel	7.85	1.41

#### Non-dimensionalization

• 
$$P/\alpha L$$
  $\alpha = \sqrt{\rho_p/\rho_t}$ 

$$\bullet \quad \left(\frac{\rho_p V^2}{\sigma_t}\right)^{0.2} \alpha^{-0.14}$$



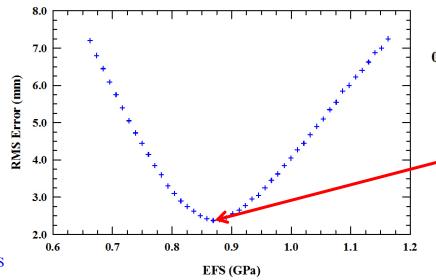
C. E. Anderson, Jr. and J.P. Riegel III, "A penetration model based on experimental data," *Int. J. Impact Engng.*, **80**: 24-35, 2015.

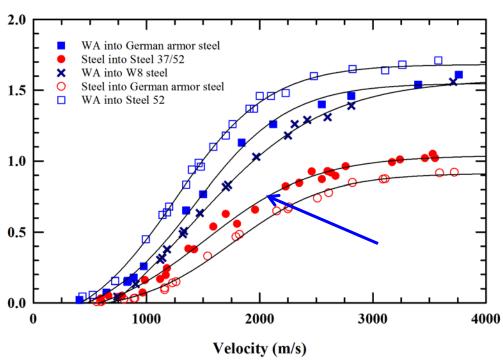
### Effective Flow Stress



#### Use Walker-Anderson model

- Parametric study on P/L as a function of velocity for various values of  $\sigma_t$
- Select value of  $\sigma_t$  that minimized the root mean square error on P/L





$$\sigma_t = 0.858 \text{ GPa}$$

$$R_t = \frac{7Y_t}{3} \ln(\alpha)$$

J.P. Riegel, III and C. E. Anderson, Jr., "Target effective flow stress calibrated using the Walker-Anderson penetration model," *Proc.* 28<sup>th</sup> *Int. Symp. Ballistics*, DESTech Publications, Inc., **2**: 1242-1253, 2014.

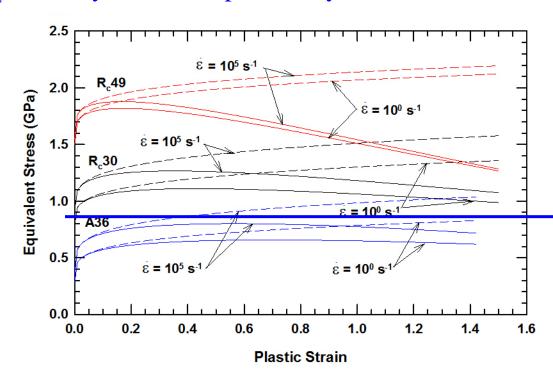


### Effective Flow Stress



#### Do these values of $\sigma_t$ have any relationship to reality?

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W8 steel	7.85	1.57
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German armor steel	7.85	1.41



Interpretation:  $\sigma_t \equiv Y_t$  is the average flow stress over the volume of the plastic zone including strain-hardening and strain-rate





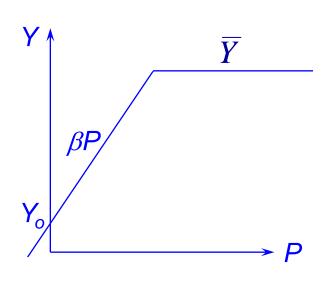
### Ceramics & Glasses



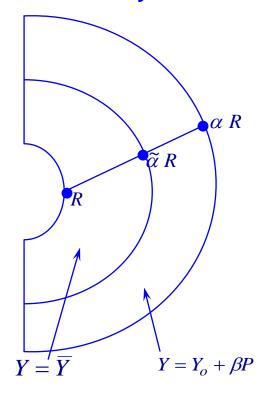
### Ceramic and Glass Penetration



Drucker-Prager yield surface



Requires solving for an interior boundary

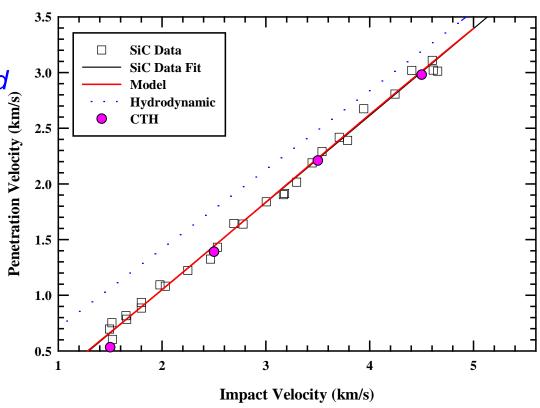




# Modeled Orphal-Franzen Experiments



- Tungsten long rods into SiC
- Assumption: penetrating failed material
- Determined  $Y_o$  (0.1 GPa),  $\beta$  (2.5), and  $Y_{cap}$  (3.7 GPa)
- Used same parameters in numerical simulations



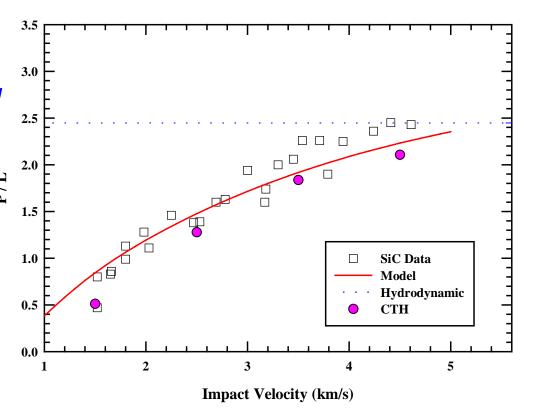
J. D. Walker, "Analytic model for penetration of thick ceramic targets," *Ceramic Transactions*, **134**: 337-348 (2002)

D. L. Orphal and R. R. Franzen, "Penetration of confined silicon carbide targets by tungsten long rods at impact velocities from 1.5 to 4.6 km/s," *Int. J. Impact Engng.*, **19**(1): 1-13 (1997).

# Modeled Orphal-Franzen Experiments



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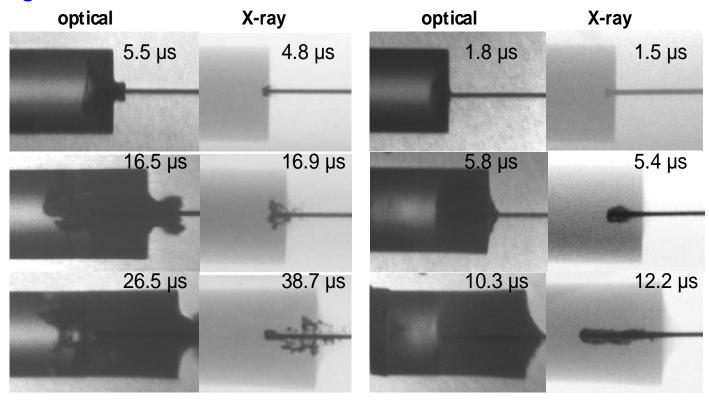
D. L. Orphal and R. R. Franzen, "Penetration of confined silicon carbide targets by tungsten long rods at impact velocities from 1.5 to 4.6 km/s," *Int. J. Impact Engng.*, **19**(1): 1-13 (1997).



### Failure Front



 Reverse ballistic experiments of borosilicate glass into a long, gold rod



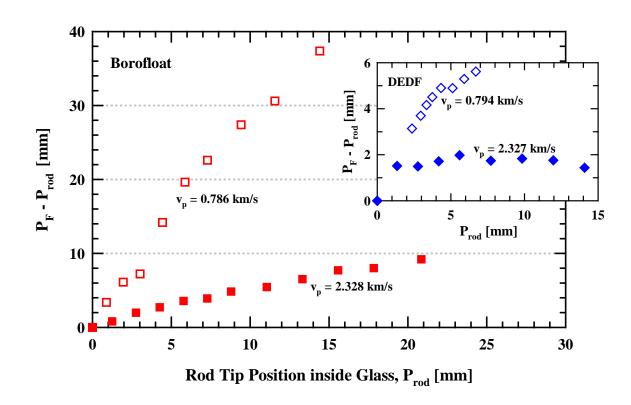
Exp. 10557,  $v_p = 786 \text{ m/s}$ 

Exp. 10585,  $v_p = 2328 \text{ m/s}$ 



# Position of Failure Front wrt Rod Tip





- Failure front moves faster than penetrating projectile
  - → projectile penetrating failed material



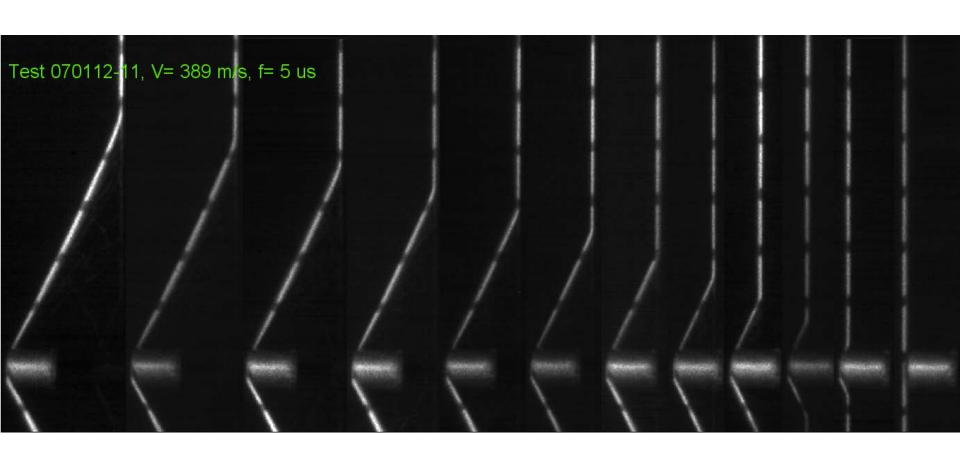


## Yarn Impact



### Yarn Impact



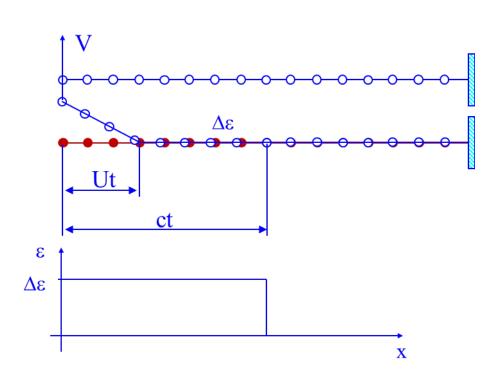




### Wave Propagation in Yarns



- Longitudinal wave travels at speed of sound c
- Transverse wave travels slower at a speed U
- Wave reflects on boundary and impact point increasing by ∆ε at each reflection until yarn breaks.



J. C. Smith, F. L. McCrackin, and H. F. Schiefer, "Stress-strain relationships in yarns subjected to rapid impact loading: Part V: Wave propagation in long textile yarns impacted transversely," *Textile Res. Journal*, **28:** 288-302, 1958.



## Wave Propagation in Yarns

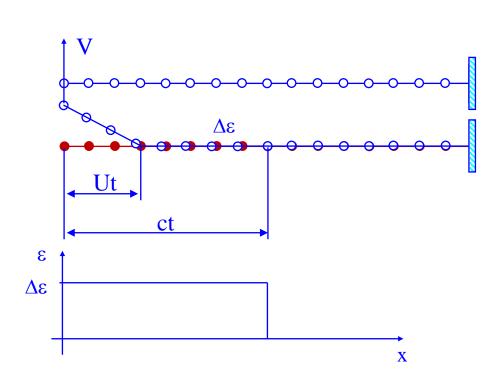


Given impact velocity and sound speed in the yarn, can determine the strain and the transverse wave velocity:

$$V = c\sqrt{\varepsilon \Big(2\sqrt{\varepsilon(1+\varepsilon)} - \varepsilon\Big)}$$

$$U = c \left( \sqrt{\varepsilon (1 + \varepsilon)} - \varepsilon \right)$$

• When  $\sum \Delta \varepsilon \geq \varepsilon_f$  yarn breaks



J. C. Smith, F. L. McCrackin, and H. F. Schiefer, "Stress-strain relationships in yarns subjected to rapid impact loading: Part V: Wave propagation in long textile yarns impacted transversely," *Textile Res. Journal*, **28:** 288-302, 1958.

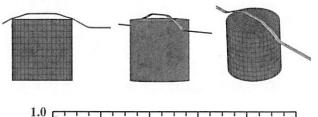


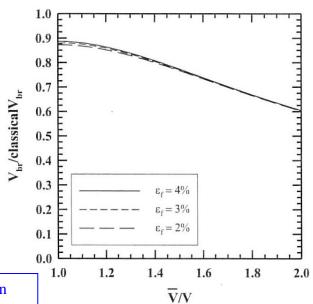
# Critical Velocity



- Critical velocity,  $V_{br}$ , is the impact velocity where the induced strain is  $\varepsilon_f$
- However, using FSP's, the critical velocity is generally less than that predicted by the classical Smith theory, e.g., ~627 m/s instead of 945 m/s for KM2 ( $\varepsilon_f$  = 4.25%)
- An analytical model was developed that incorporates the wave interactions from the sides of the flat projectile, plus any "bounce" of the yarn, reduces the critical velocity

$$V_{br} = c\sqrt{\varepsilon_f \left(2\sqrt{\varepsilon_f \left(1 + \varepsilon_f\right)} - \varepsilon_f\right)}$$





J. D. Walker and S. Chocron, "Why impacted yarns break at lower speed than classical theory predicts," *J. Appl. Mech.*, **78**: 051021-1/7, 2011.





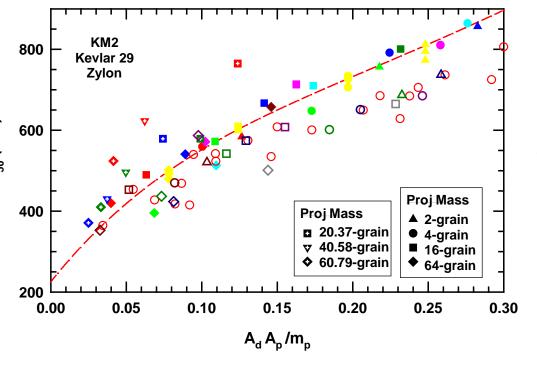
# Fabric Response & Resin-Impregnated Fabrics



# Response of Fabrics



- Like to plot  $V_{50}$  as some function that accounts for different fabric materials, different number of plies, and different projectile masses
- Defined a nondimensional abscissa:
- A<sub>d</sub> = areal density of fabric (accounts for different number of plies and fabric density)
- m<sub>p</sub>/A<sub>p</sub> = areal density of projectile (accounts for different projectile cross-sectional areas and projectile masses)



P. M. Cunniff, "Dimensionless parameters for optimization of textile-based body armor systems," *Proc. 18th Symp. Ballistics*, **2**: 1303-1310, Technomic Publishing Co, Lancaster, PA, 1999.

Red 12 Green 20 Yellow 23 Blue 27 Pink 30 Cyan 33

Gray

KM2/#Plies

Kev29/#Plies D. Red 8 D. Green 16 D. Blue 18

D. Pink 22 D. Gray 54

Zylon/#Plies Blue 28 D. Green 37 Red 46



# Fabric Response



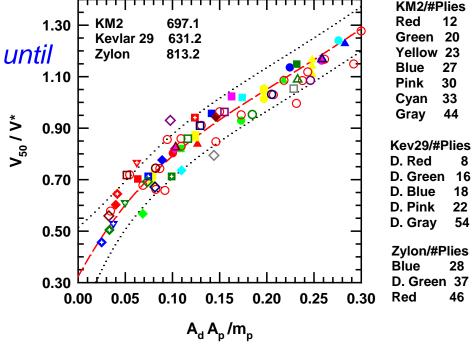
Couple fabric response to properties of the yarns

 Typically, yarns are linearly elastic until failure

Strain energy per volume:

$$U = \frac{1}{\rho} \int \sigma d\varepsilon = \frac{1}{2} \frac{\sigma_u}{\varepsilon_f}$$

- Cunniff defined a new variable
- $U^* = Uc = U\sqrt{E/\rho} \text{ [m}^3/\text{s}^3\text{]}$
- $V^* = (U^*)^{1/3}$



P. M. Cunniff, "Dimensionless parameters for optimization of textile-based body armor systems," *Proc. 18th Symp. Ballistics*, **2**: 1303-1310, Technomic Publishing Co, Lancaster, PA, 1999.

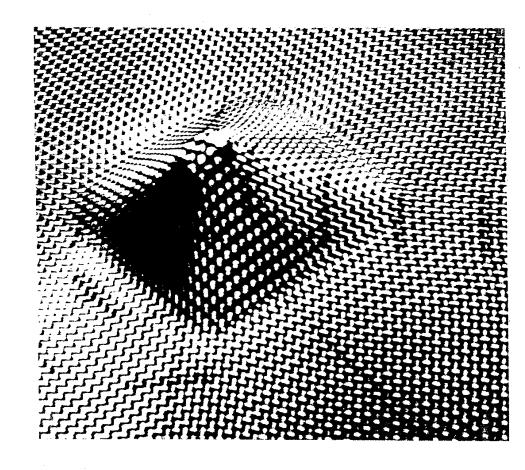


# Fabric Modeling



113

- Deformation of fabrics results in a definitive pyramidal shape
- Modeling this response analytically was a challenge

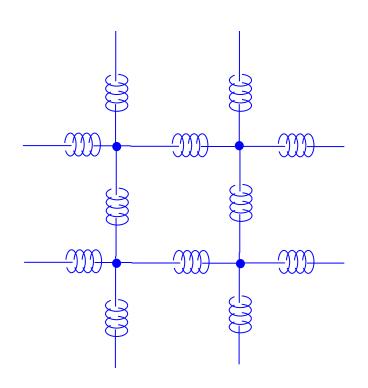




# Fabric Modeling



- Fabric a limiting case of a spring system
  - Solution to static deflection of fabric sheet
  - Determine strains from sheet deflection
  - Use the strain to determine force versus deflection
  - Use Piola-Kirhhoff stress (stress with respect to the initial configuration)



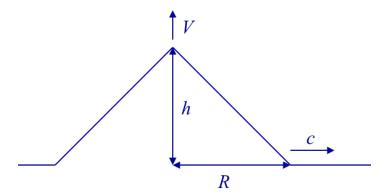
J. D. Walker, "Constitutive model for fabrics with explicit static solution and ballistic limit," *Proc.* 18th Int. Symp. Ballistics, 2: 1231-1238, Technomic Publishing Co., Lancaster, PA, 1999.



# Out-of-Plane Solution



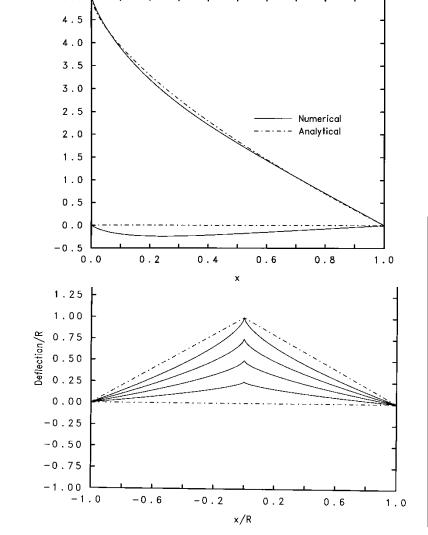
Develop solution for out-ofplane deformation



$$u_z = h \left\{ 1 - \sqrt{(x/R)^{4/3} + (y/R)^{4/3}} \right\}$$

$$h \left\{ 1 - (y/R)^{2/3} \right\}$$

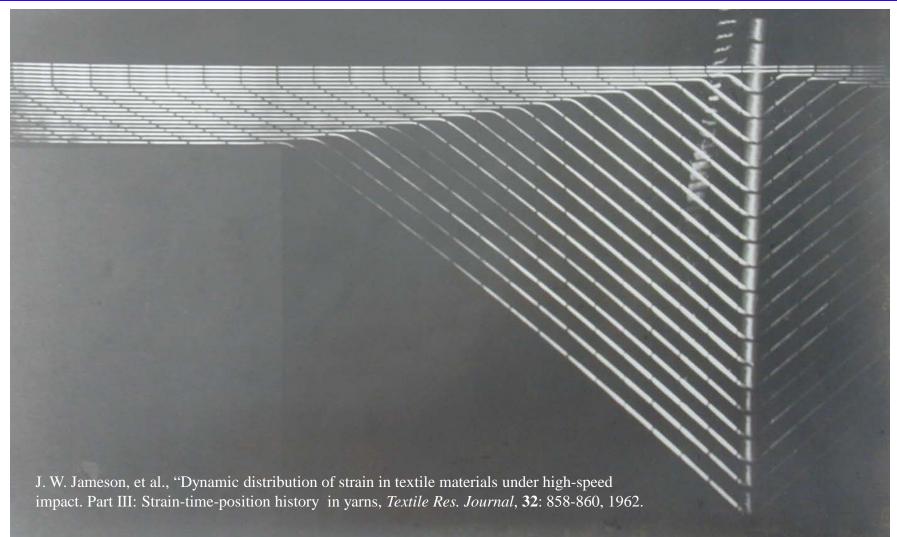
$$\varepsilon_x = \frac{2}{9} (h/R)^2 (R/x)^{2/3}$$





# Yarn Impact (Jameson, 1957)







#### In-Plane Motion



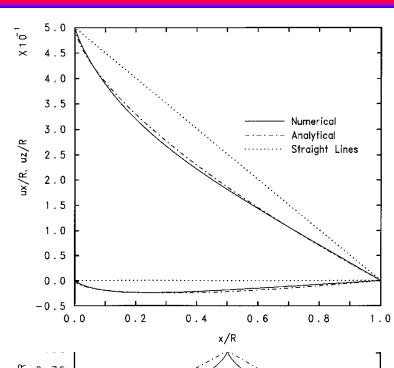
#### Developed an equation for in-plane motion

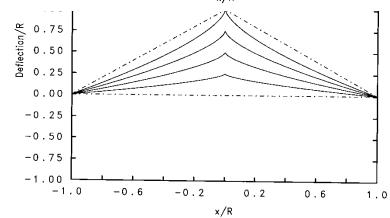
$$u_{x}(x,y) = -\frac{2}{3} \frac{h^{2}}{R} \left\{ 1 - (y/r)^{4/3} \right\}^{1/4}$$

$$\left\{ \left( 1 - (y/R)^{2/3} \right) \left[ 1 - \frac{x/R}{\left\{ 1 - (y/R)^{2/3} \right\}^{3/4}} \right] - \left( 1 - \sqrt{(x/R)^{4/3} + (y/R)^{4/3}} \right) \right\}$$

$$u_{x} = -\frac{2}{3} \frac{h^{2}}{R} \left\{ \left( \frac{x}{R} \right)^{2/3} - \frac{x}{R} \right\}$$

J. D. Walker, "Constitutive model for fabrics with explicit static solution and ballistic limit," *Proc. 18th Int. Symp. Ballistics*, **2**: 1231-1238, Technomic Publishing Co., Lancaster, PA, 1999.

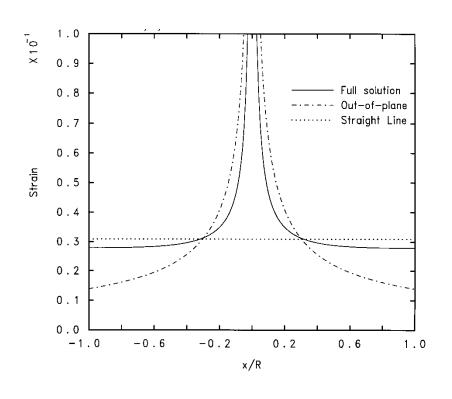






# Strain Along Edge





- Analytic solution allows calculation of strain
- Strain along edge

$$\varepsilon_{x} \approx \frac{2}{9} \left( \frac{h}{R} \right)^{2} \left\{ \left( \frac{R}{x} \right)^{2/3} - 2 \left( \frac{R}{x} \right)^{1/3} + 3 \right\}$$

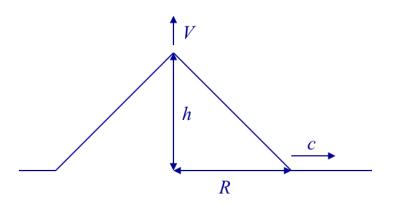
 Strains from out of plane solution differ from total solution

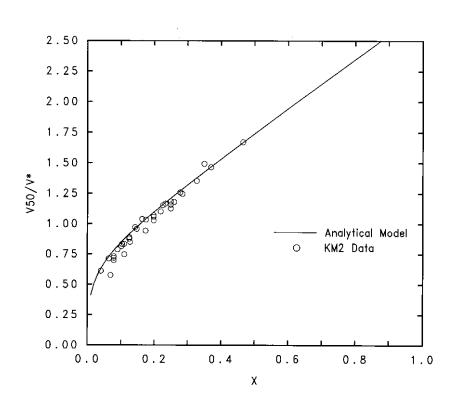


# Calculate V<sub>50</sub>



- Need to know how fast the tent is expanding under the impact (how much fabric is involved in the impact)
- Set deceleration of projectile and fabric equal to the force



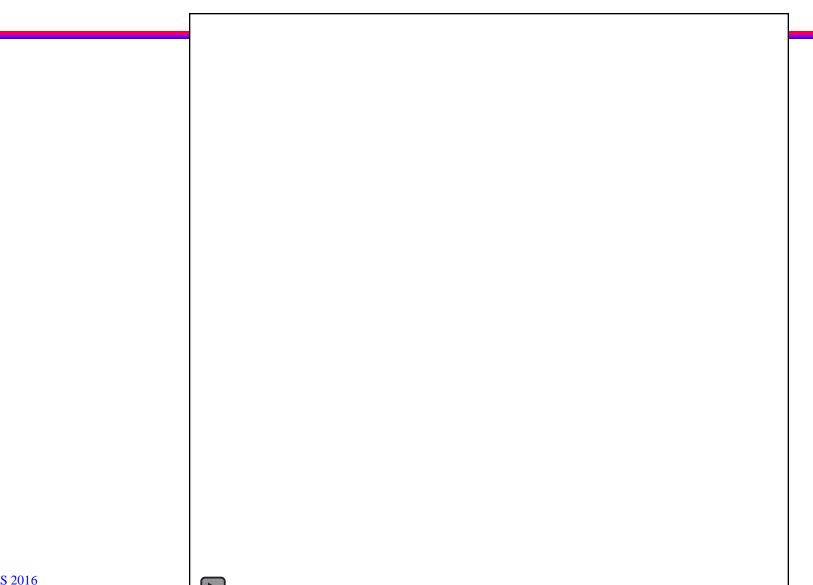


J. D. Walker, "Constitutive model for fabrics with explicit static solution and ballistic limit," *Proc. 18th Int. Symp. Ballistics*, **2**: 1231-1238, Technomic Publishing Co., Lancaster, PA, 1999.



# Fabric Model



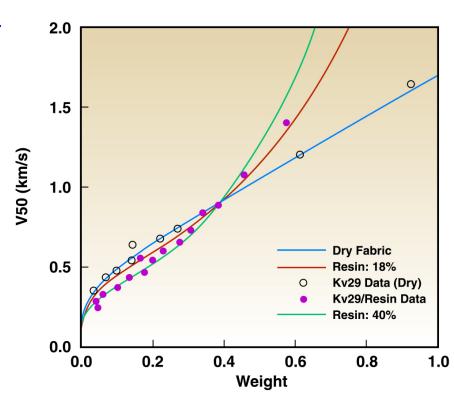




### Addition of Resin to Fabrics



- For same areal density, addition of resin means removal of fabric (loss of tensile strength)
- Resin adds bending moment to response, thus increasing strength
- Addition of resin increases shear wave speed, thus reducing the strain
- Harder composite deforms projectile
- Fabrics held in resin may now shear







# Dwell and Dwell-Penetration Transition



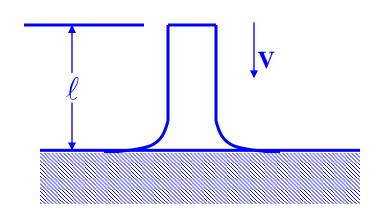


# An Analytic Model for Dwell



$$\rho_p \ell \, \frac{d\mathbf{v}}{dt} = -Y_p$$

$$\frac{d\ell}{dt} = -(\mathbf{v} - \mathbf{u})$$



for dwell u = 0

C. E. Anderson, Jr. and J. D. Walker, "An analytic model for dwell and interface defeat," *Int. J. Impact Engng.*, **31**(9): 1119-1132 (2005)



# Analytic Dwell Model-1



$$\rho_p \ell \frac{d\mathbf{v}}{dt} = -Y_p, \qquad \frac{d\ell}{dt} = -\mathbf{v}$$

Can be solved simultaneously and integrated:

$$\frac{1}{2}\rho_p\left(\mathbf{v}^2 - \mathbf{v}_o^2\right) = Y_p \ln\left(\ell/\ell_o\right)$$

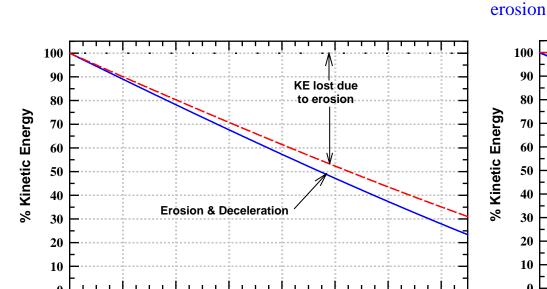
After some rearranging

$$KE = \frac{1}{2} \rho_p \ell v^2 = \frac{1}{2} \rho_p (\ell - \ell_0) v_0^2 + Y_p \ell \ln(\ell / \ell_0) + KE_0$$

# Analytic Dwell Model-2



$$\frac{1}{KE_{0}} \frac{d KE}{dt} = -\frac{\mathbf{v}}{\ell_{0}} \left\{ 1 + \frac{Y_{p}}{\frac{1}{2} \rho_{p} \mathbf{v}_{0}^{2}} \ln(\ell/\ell_{0}) + \frac{Y_{p}}{\frac{1}{2} \rho_{p} \mathbf{v}_{0}^{2}} \right\}.$$



0.3

0.4

 $t \, \mathbf{v_0} / L_0$ 

0.2

0.1

0.0

100 90 KE lost due 80 to erosion % Kinetic Energy 70 **60 50** KE lost due **KE with Mass Loss** to velocity decay (Erosion Only) **30 Erosion & Deceleration** 20 10 10 15 20 25 **30** 5 Time (□s)

deceleration

Long Rod

0.6

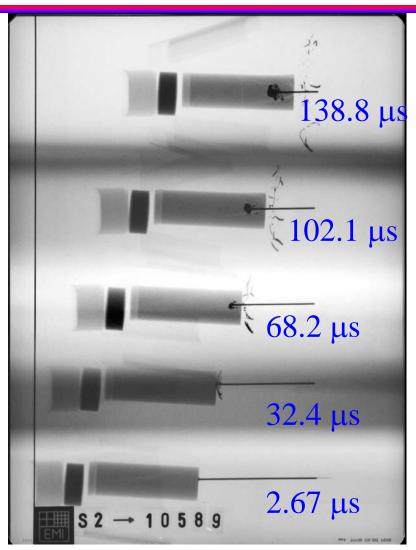
0.5

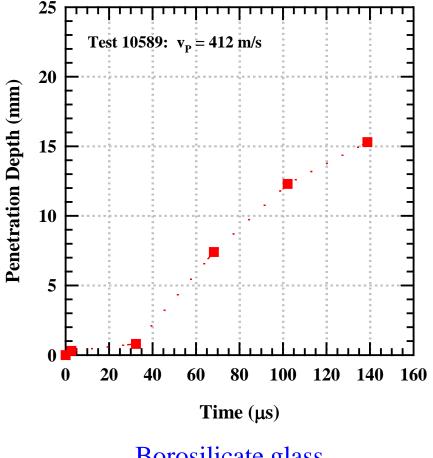
0.7



#### **Dwell-Penetration Transition**







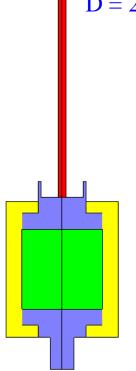


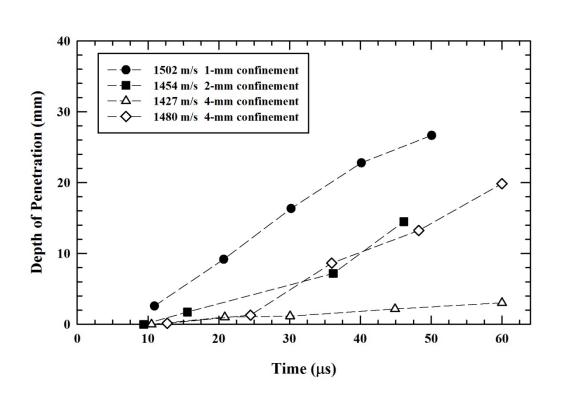
# Confined $B_4C$ Experiments





D = 2 mm





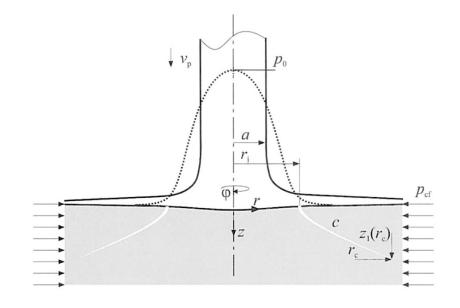
L. Westerling, P. Lundberg, and B. Lundberg, "Tungsten long-rod penetration into confined cylinders of boron carbide at and above ordnance velocity," *Int. J. Impact Engng.*, **25**(7): 703-714, 2001.



# Transition from Interface Defeat



- Lundberg and colleagues have been studying interface defeat for a number of years (1998-present)
- Modeled the pressure distribution, P, of the projectile on the surface
  - Radial distribution from low-velocity water jet results
  - α << 1; ratio of elastic to inertial effects
  - β << 1; ratio of plastic to inertial effects
  - $K_p$  = bulk modulus;  $q_p$  = Bernoulli pressure;  $V_o$  = impact speed



$$P(r,\alpha,\beta) = q(r) \left[ 1 + \frac{1}{2\alpha} + 3.0\beta - 1.6\beta^2 \right]$$

$$\alpha = \frac{K_p}{q_p} \quad \beta = \frac{\sigma_{yp}}{q_\rho} \quad q_p = q(0) = \frac{1}{2} \rho_p V_o^2$$

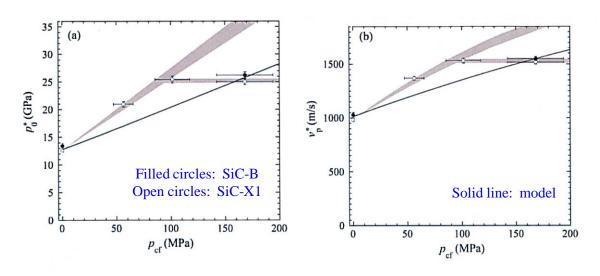
P. Lundberg, R. Renström, and O. Andersson, "Influence of length scale on the transition from interface defeat to penetration in unconfined ceramic targets," *J. Appl. Mech.*, **80**: 031801-1/9, 2013.



# Transition from Interface Defeat Swill



- First estimated transition velocity by equating the maximum load per unit area with the shear yield strength of the ceramic material
- Subsequent work used fracture mechanics to estimate the critical stress, P<sup>\*</sup><sub>0</sub> to drive a crack, first for an unconfined target, then a target with applied prestress, P<sub>cf</sub>



Solid line: model estimate Gray shading: possible twomode behavior

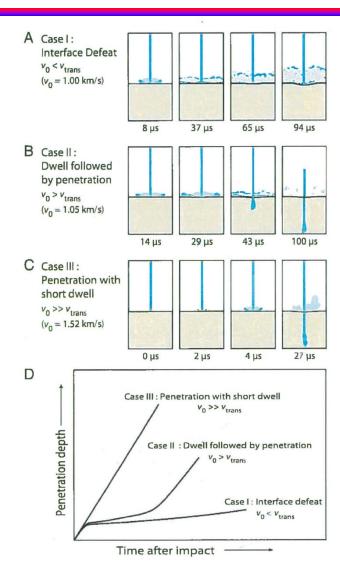
P. Lundberg, R. Renström, and O. Andersson, "Influence of confining prestress on the transition from interface defeat to penetration in ceramic targets," *Defence Technology*, in press, 2016.



# Three Regimes of Penetration



- Case 1.  $V_o < V_T$ : Interface defeat
- Case 2. V<sub>o</sub> > V<sub>T</sub>: Dwell followed by penetration
- Case 3. V<sub>o</sub> >> V<sub>T</sub>: Short dwell followed by penetration



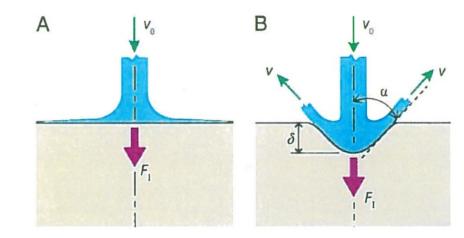
T. Uth and V. S. Deshpande, "Unsteady penetration of a target by a liquid jet," *PNAS*, **110**(50): 20028-33, 2013



# Fluid-Structure Interaction Model



- Unsteady penetration rate due to impact of a fluid jet with velocity V<sub>o</sub>
- During initial stages of jet impact, the target surface is flat and the fluid spreads horizontally
- As the jet deforms the target and penetrates at depth  $\delta$ , it creates a dimple at the impact site
- Flow pattern changes, resulting in backflow of the fluid with a velocity V and a consequent increase in the penetration pressure



$$P = \frac{F_I}{A_{jet}} = (1 + \cos \alpha) \rho_{jet} V_o^2$$

T. Uth and V. S. Deshpande, "Unsteady penetration of a target by a liquid jet," *PNAS*, **110**(50): 20028-33, 2013



# Experiments: 2-mm-diameter Water Jet on Vacuum Grease



#### Provided experimental evidence

- Case 1. V<sub>o</sub> < V<sub>T</sub>: Pressure remains constant; radially flow of jet
- Case 2.  $V_o > V_T$ : Pressure increases sharply at  $t \sim 500 \ \mu s$ ;  $\delta \approx r_{jet}$  penetration rate increases rapidly
- Case 3.  $V_o >> V_T$ : Measurements and observations closely resemble those of Case 2, but with no discernible dwell phase

Case I Pressure  $(v_n = 6.5 \text{ ms}^{-1})$ 0.2  $t = 100 \, \text{ms}$  $t = 1100 \, \text{ms}$ 400 600 800 1000 1200 Time after impact t (ms) B Case II d  $v_{0} > v_{trans}$ ( $v_{0} = 7.0 \text{ m s}^{-1}$ ) Pressure 0.5 100 200 300 400 500 600 700 Time after impact t (ms) t = 515 ms (kPa) Case III d  $v_{c} >> v_{trans}$   $(v_{c} = 9.0 \text{ ms}^{-1})$ 0.5 Time after impact t (ms)

T. Uth and V. S. Deshpande, "Unsteady penetration of a target by a liquid jet," *PNAS*, **110**(50): 20028-33, 2013

$$P = \frac{F_I}{A_{jet}} = (1 + \cos \alpha) \rho_{jet} V_o^2$$



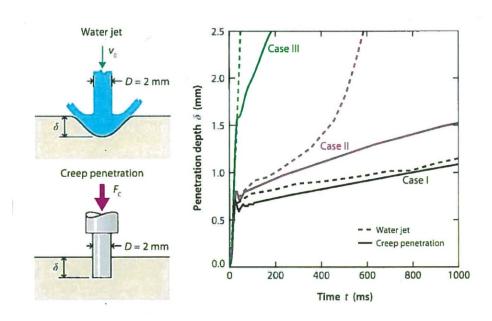
# Jet vs. Creep Penetration



$$P_{creep} = \rho_{jet} V_o^2$$

$$P = \frac{F_I}{A_{jet}} = (1 + \cos \alpha) \rho_{jet} V_o^2$$

- Case 1: 2 experiments give similar response creep
- Case 2: Penetration rate for creep case remains steady; sharp increase in jet penetration rate
- Case 3: Backflow sets in early and penetration rate is higher for the jet compared to creep experiment



T. Uth and V. S. Deshpande, "Unsteady penetration of a target by a liquid jet," *PNAS*, **110**(50): 20028-33, 2013



# Fluid-Structure Interaction and Dwell-Penetration Transition



- Demonstrated that backflow of impacting jet, rather than any damage to the target material, can cause unsteady penetration
- Experimentally, damage observed in ceramics that show interface defeat
- Damage and backflow are not mutually exclusive
- Impacting jet needs to penetrate to a depth of approximately the jet radius before backflow can be established
- Brittle targets such as ceramics must have some damage to permit penetration to half of the projectile radius
- Backflow then acts like a switch, doubling the pressure, and amplifying the penetration rate





# Summary



# *Modeling*



- Understanding is displayed by how well we can model
- Accuracy versus trends:
  - many difficult problems, accuracy is subjective
  - but if cannot get trends right, then do not understand
- Modeling
  - materials response (constitutive) models
  - numerical simulations
  - analytical models



# **Modeling**



- If you can model the phenomenology, it demonstrates a certain level of understanding
- Of course, we have to be careful that we are truly modeling, and not simply curve fitting (adjusting parameters)
- That's why in penetration mechanics modeling, need to verify the ability to predict at different velocities, different geometries, and different materials

If have numerical simulations, why the need for analytical models?



# Analytical Modeling



- If we can develop an analytical model that captures the essence of the phenomenology:
  - Not only demonstrated that we understand
  - Also demonstrates that we have grasped the essential and relevant mechanics of the phenomenology
- Now have a tool for predictions, design studies, optimization, etc., that is fast running compared to numerical simulations

IBS 2016 138





# THE END